

# SiC Power Modules for More-Electric Aircraft Application

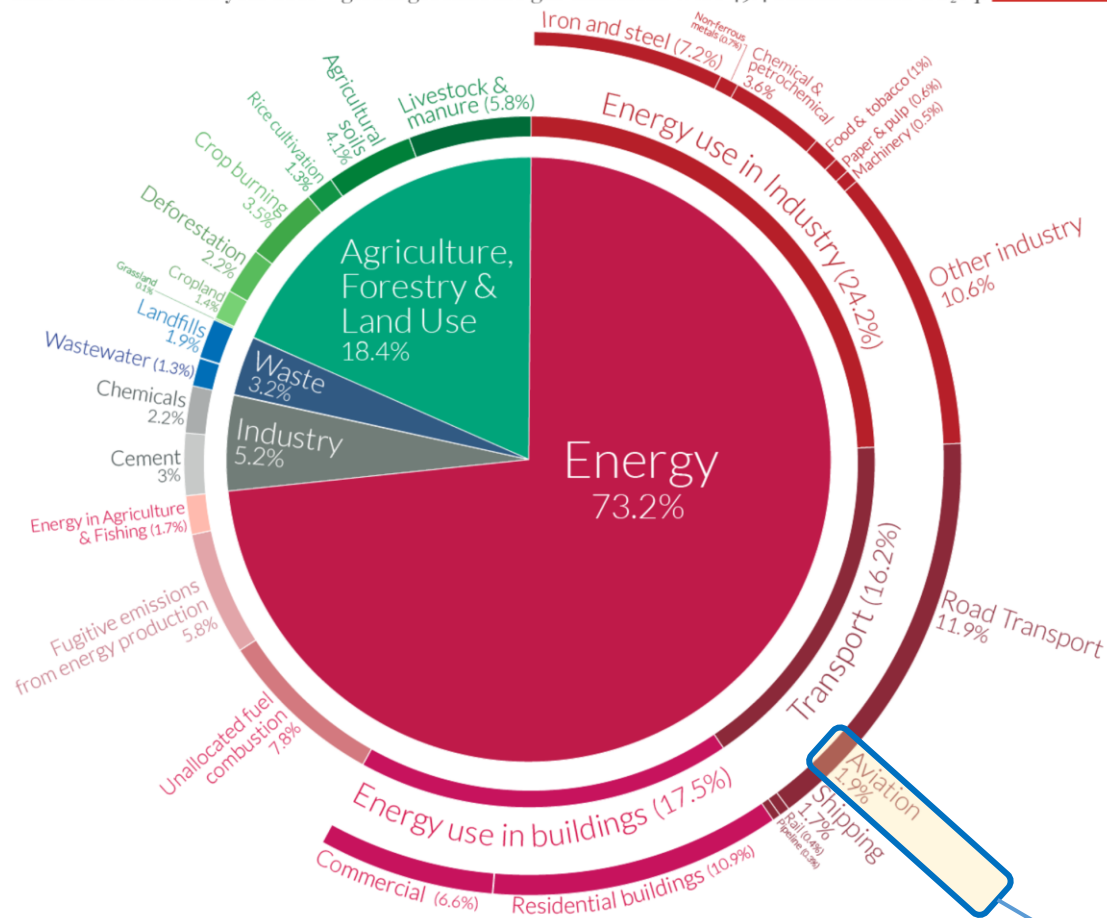
**Antonio Griffo**

The University of Sheffield, UK

# The More-Electric Aircraft

## Global greenhouse gas emissions by sector

This is shown for the year 2016 – global greenhouse gas emissions were 49.4 billion tonnes CO<sub>2</sub>eq.

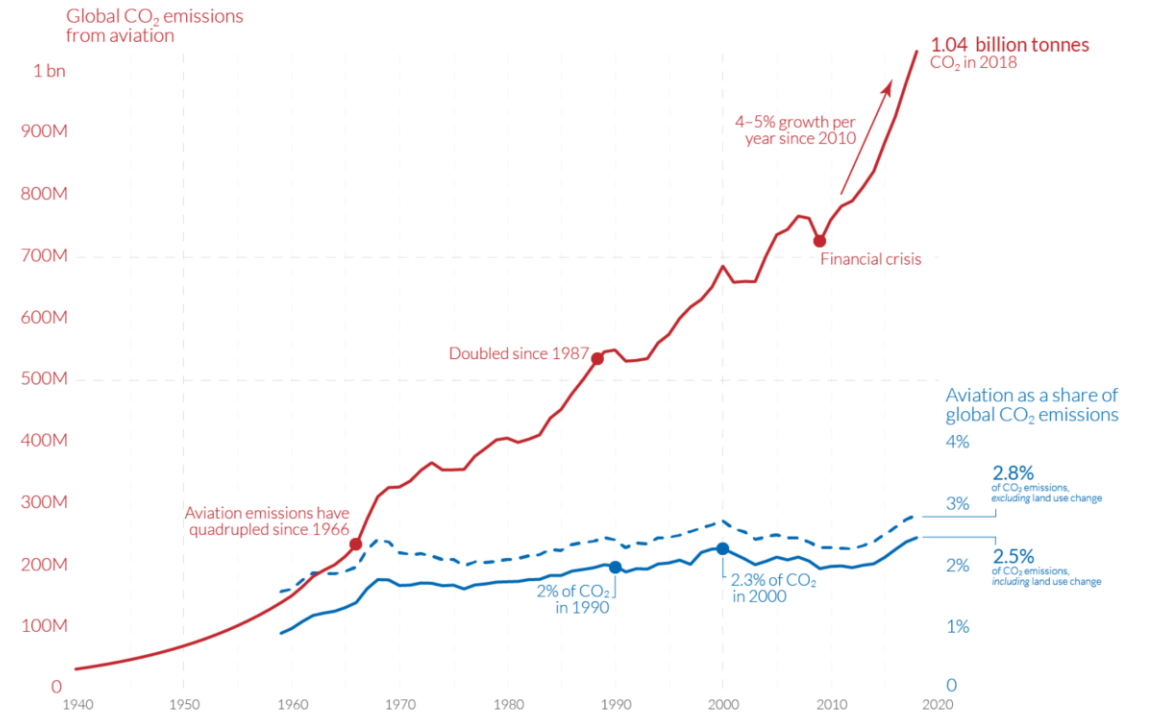


OurWorldinData.org – Research and data to make progress against the world's largest problems.  
Source: Climate Watch, the World Resources Institute (2020).

Licensed under CC-BY by the author Hannah Ritchie (2020).

## Global carbon dioxide emissions from aviation

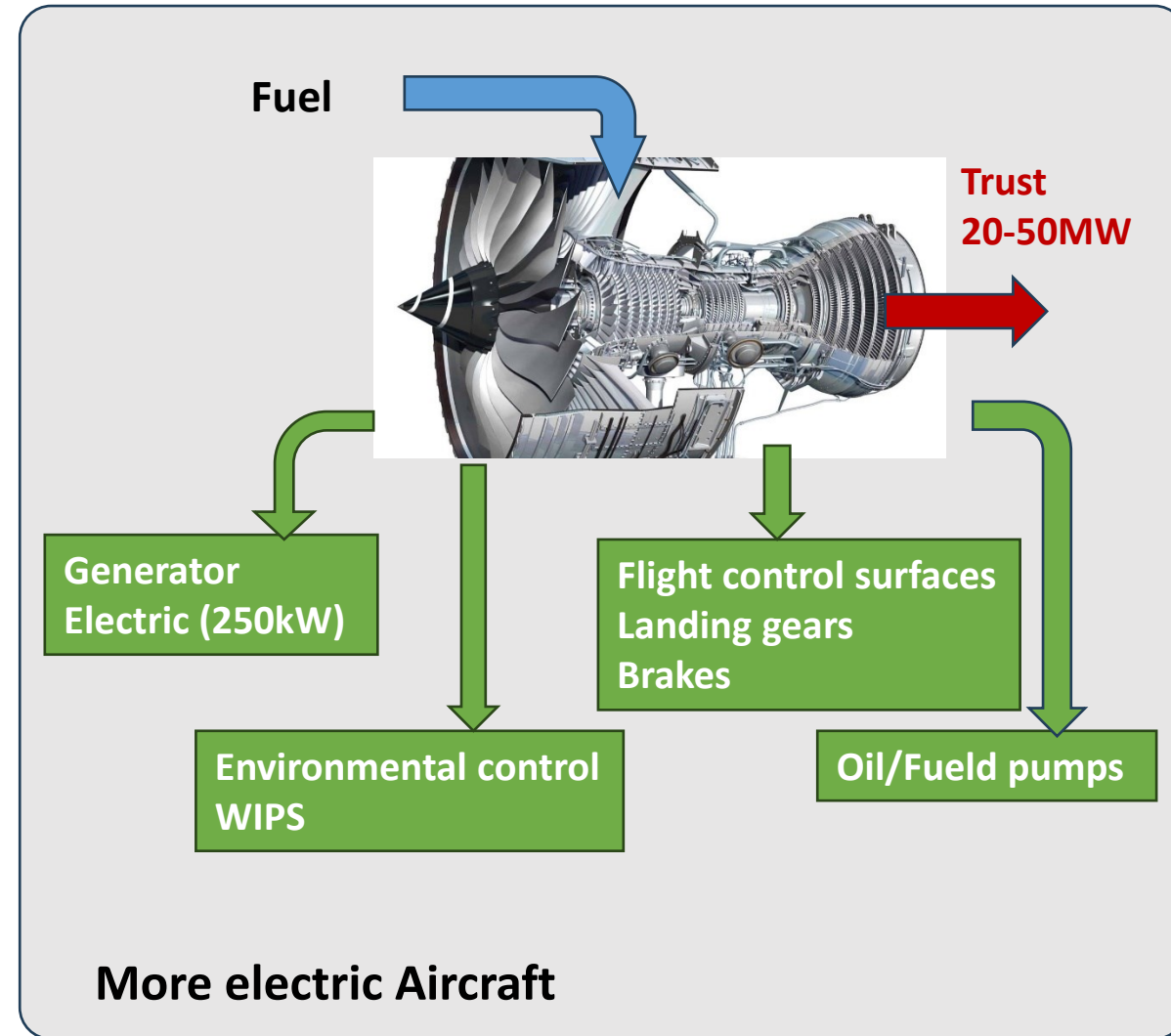
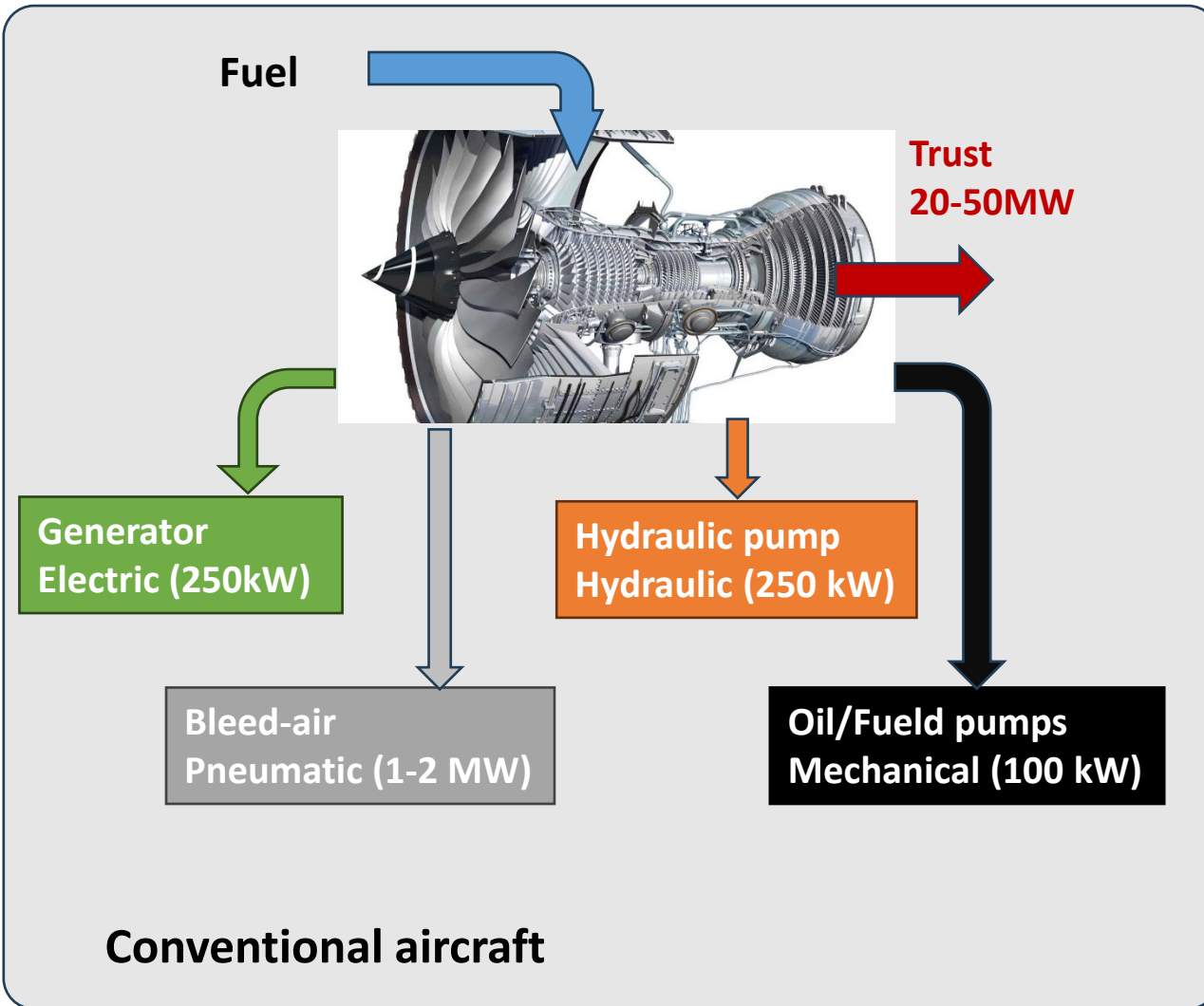
Aviation emissions includes passenger air travel, freight and military operations. It does not include non-CO<sub>2</sub> climate forcings, or a multiplier for warming effects at altitude.



OurWorldinData.org – Research and data to make progress against the world's largest problems.  
Source: Lee et al. (2020). The contribution of global aviation to anthropogenic climate forcing for 2000 to 2018; based on Sausen and Schumann (2000) & IEA. Share of global emissions calculated based on total CO<sub>2</sub> data from the Global Carbon Project. Licensed under CC-BY by the author Hannah Ritchie.

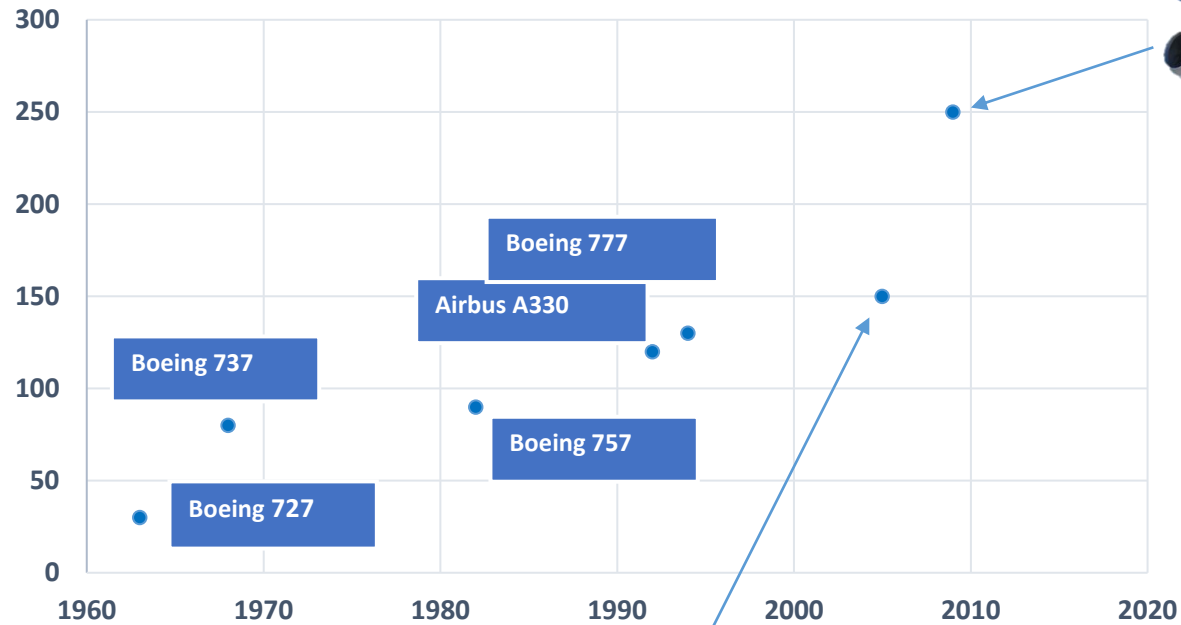
**Aviation accounts for 1.9% global CO<sub>2</sub> emissions**

# The More-Electric Aircraft



# The More-Electric Aircraft

Single Generator Rating [kVA]



## Boeing 787 (up to 410pax)

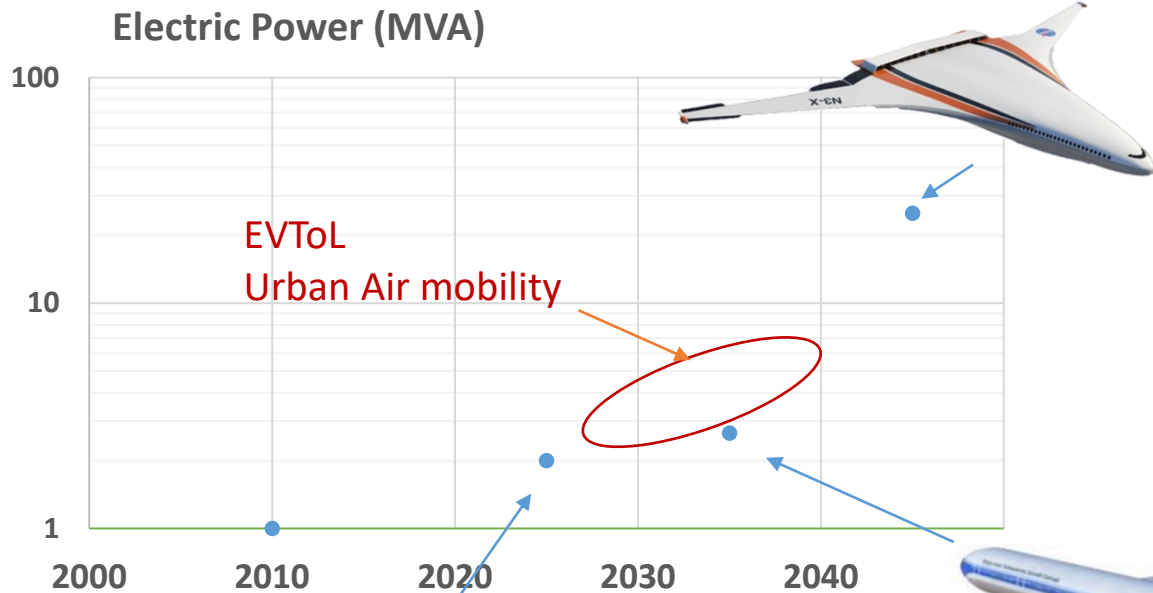
- large number of pneumatic and hydraulic systems replaced by Power electronics and electrical machines
- compressors replace pneumatic loads, such as the WIPS and the ECS.
- AC voltage 235 V
- Four generators on the two main engines generating a total electrical power of 1 MVA
- $\frac{P_{electrical}}{P_{thermal}} < 1.5 \%$



## Airbus A380 (up to 853 seats)

- Hydraulic drives are partly electrified (flight control systems)
- AC voltage 115V
- Four generators on the main engines, total elect. 600 kVA
- $\frac{P_{electrical}}{P_{thermal}} < 0.2 \%$

# The More-Electric Aircraft



## NASA N3-X (300 pax)

- Superconducting machines
- VDC= ±2 kV
- Two generators, 6.5 MW each per wing.
- Sixteen distributed electric drive systems provide the thrust
- Nominal total electrical output 25 MW
- $\frac{P_{electrical}}{P_{thermal}} = 80\%$

## NASA Single-Aisle Turboelectric Aircraft with Aft Boundary Layer Propulsion (STARC-ABL) (2030-35)

154 pax

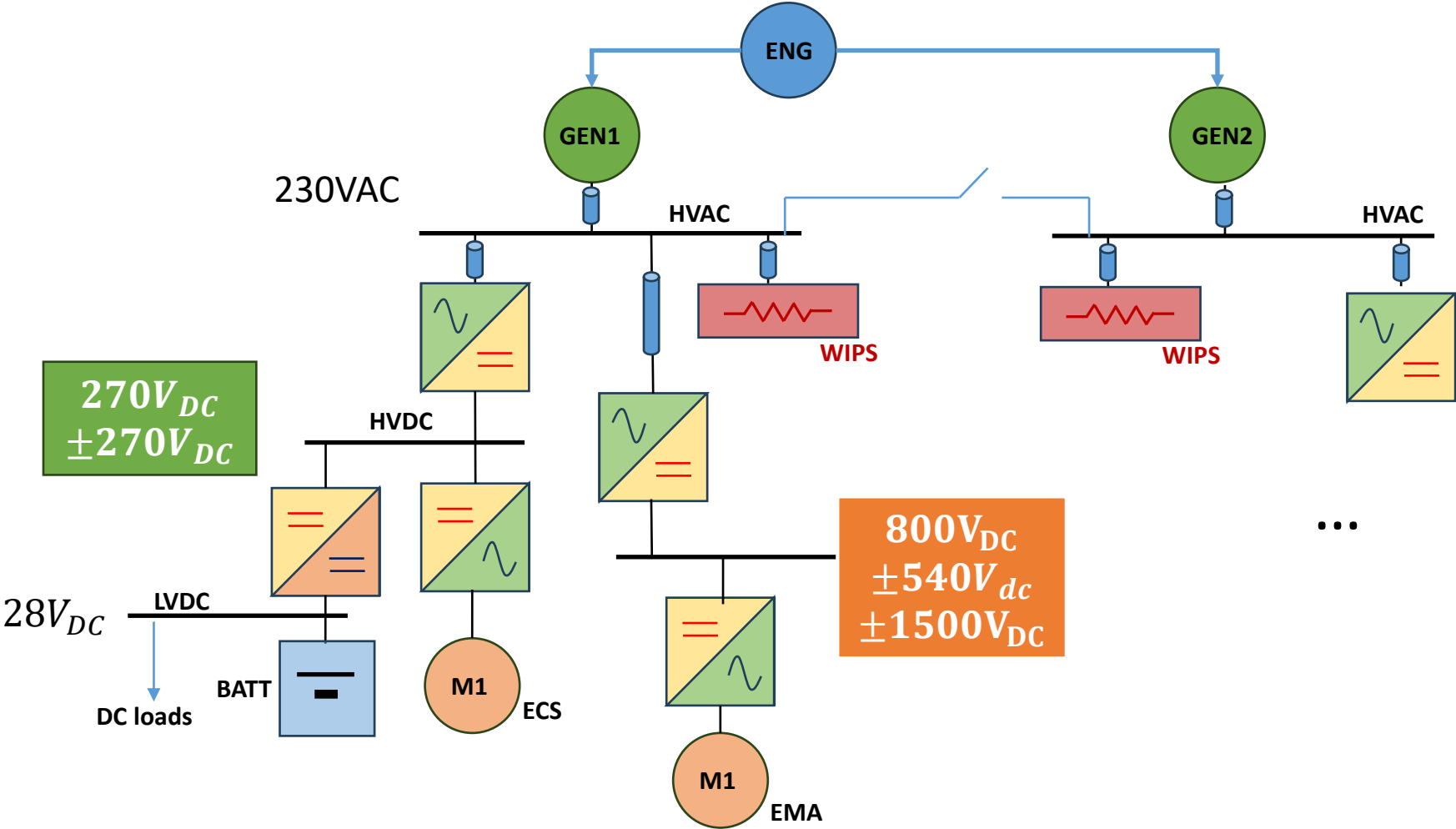
- Two engine, electrical power of 1.5 MW
- 1 kV DC on-board power supply
- boundary layer ingesting tail-one thruster
- All-electric drive, power of 2.61 MW and generates thrust
- $\frac{P_{electrical}}{P_{thermal}} = 30\%$



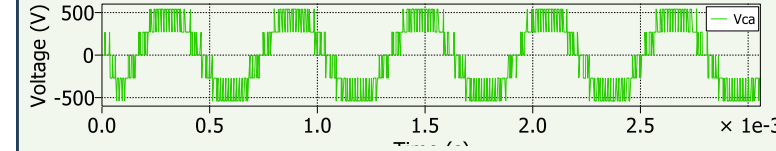
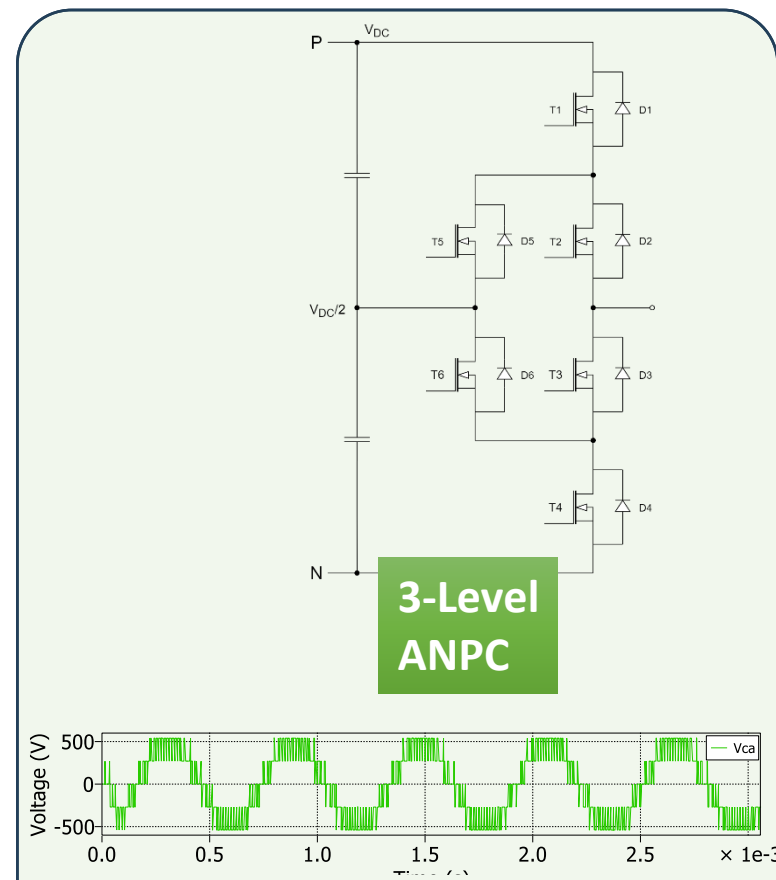
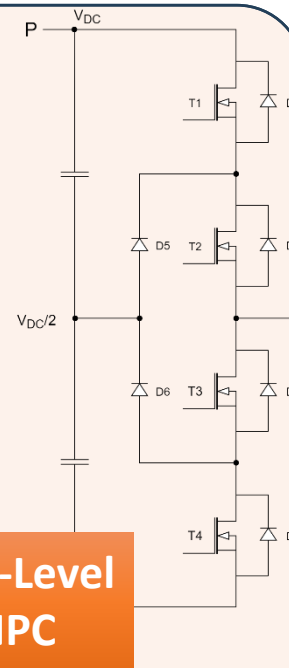
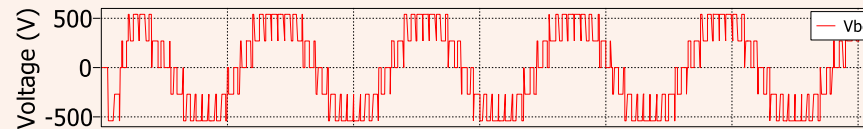
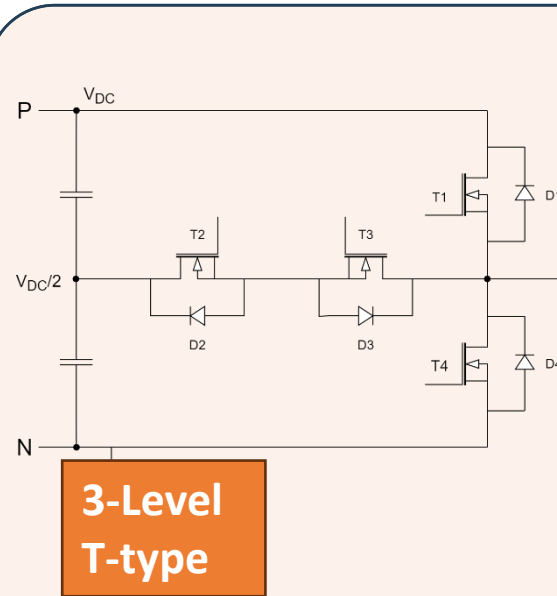
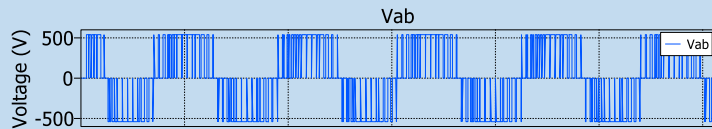
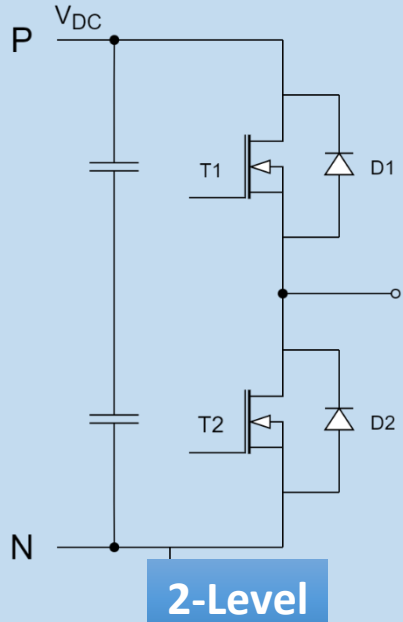
## E-FAN X

- BAe 146 powered by four engines.
- One propulsion unit powered by a 2 MW
- Vdc=3kV
- $\frac{P_{electrical}}{P_{thermal}} = 25\%$

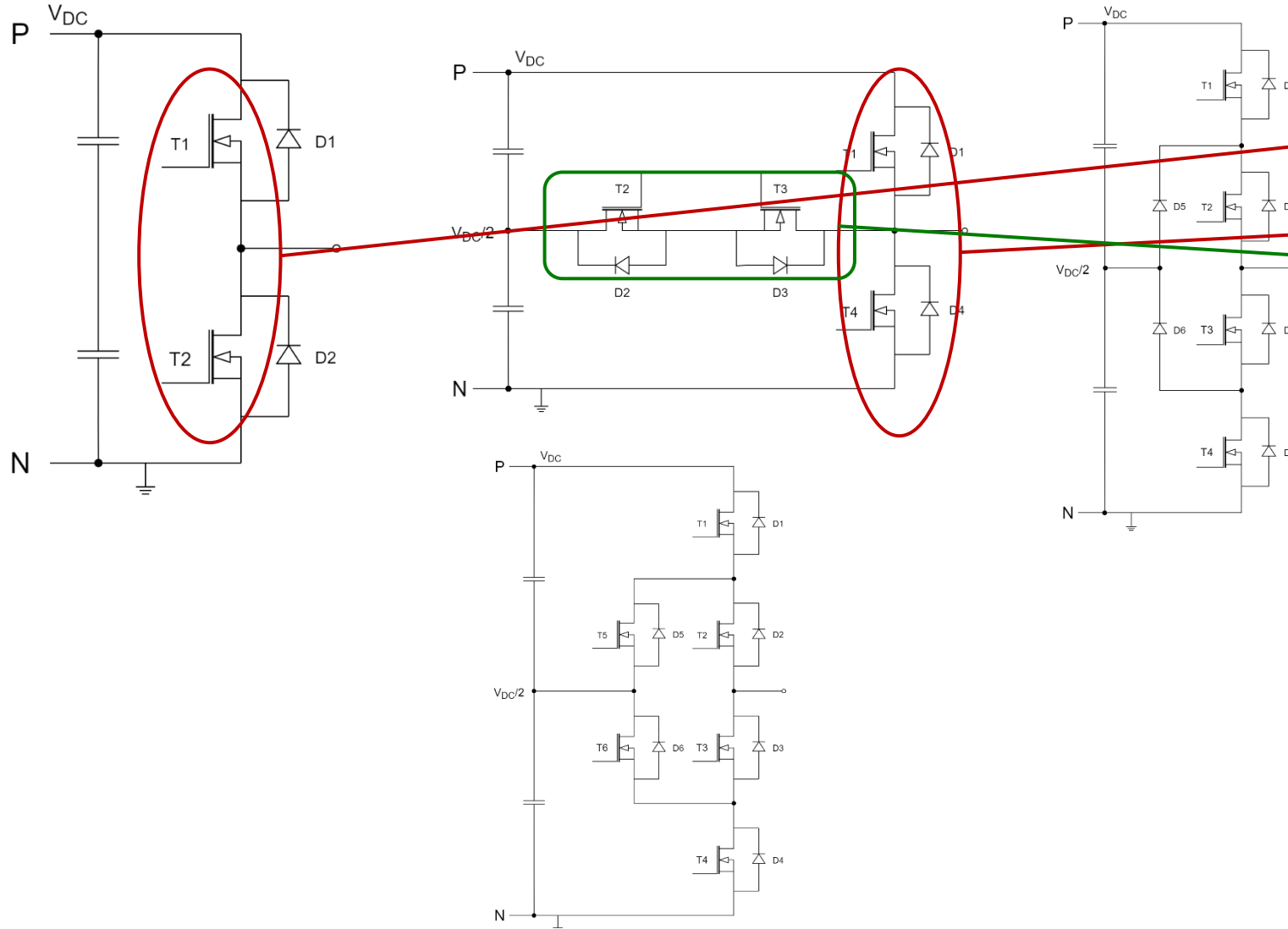
# The More-Electric Aircraft



Power electronics is a critical enabling technology for aviation electrification

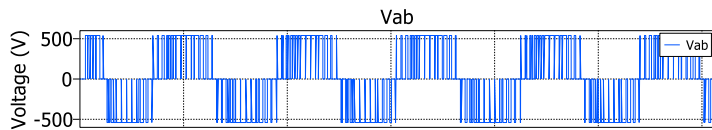
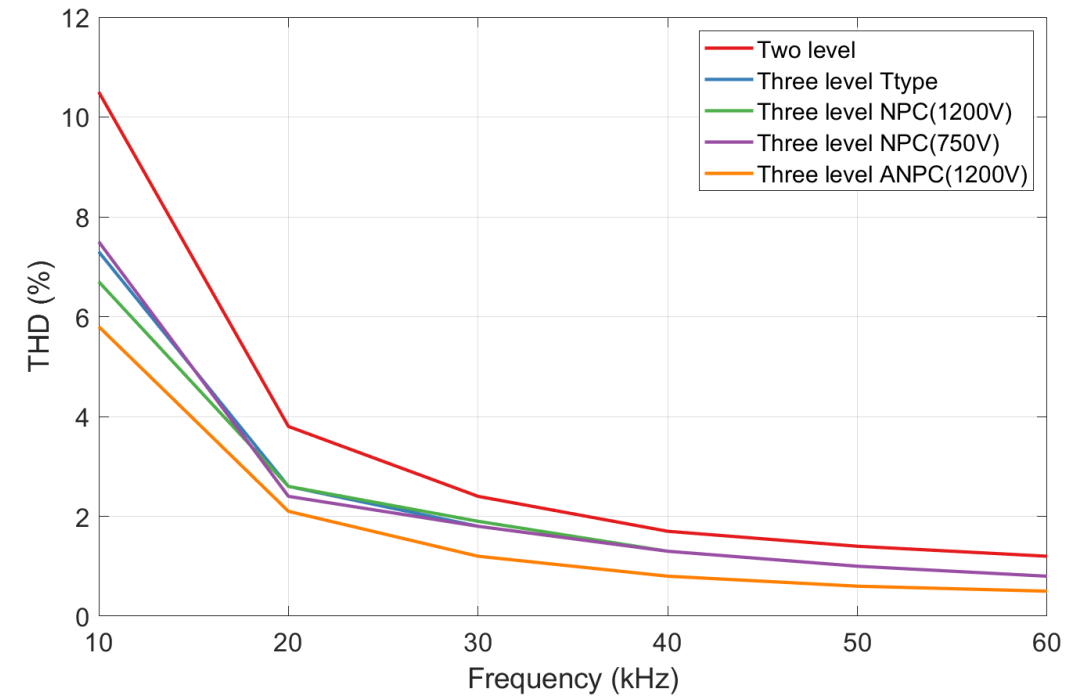
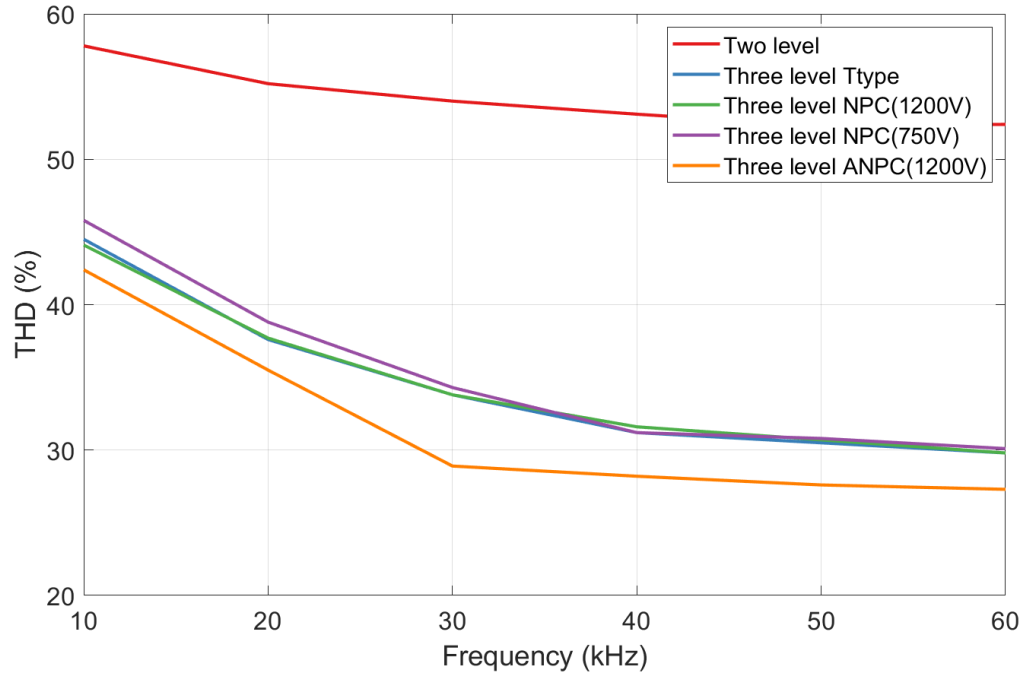


# Converters for MEA

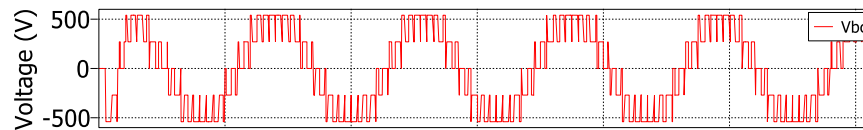


	Switch	Blocking Voltage	Switching Voltage
2-Level	T1	$V_{DC}$	$V_{DC}$
	T2		
3-Level T-Type	T1	$V_{DC}$	$\frac{1}{2} V_{DC}$
	T2		
	T3		
	T4		
3-Level NPC	T1	$\frac{1}{2} V_{DC}$	$\frac{1}{2} V_{DC}$
	T2		
	T3		
	T4		
	D5		
	D6		
3-Level ANPC	T1	$\frac{1}{2} V_{DC}$	$\frac{1}{2} V_{DC}$
	T2		
	T3		
	T4		
	T5		
	T6		

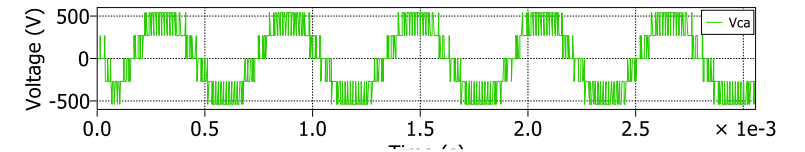




Two level



Three-level T-type & NPC



Three-level ANPC

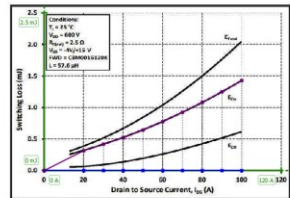


Figure 23. Clamped Inductive Switching Energy vs. Drain Current ( $V_{ds} = 600V$ )

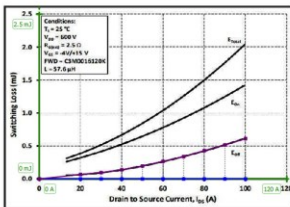


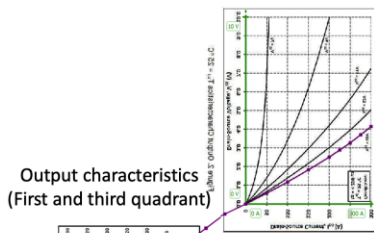
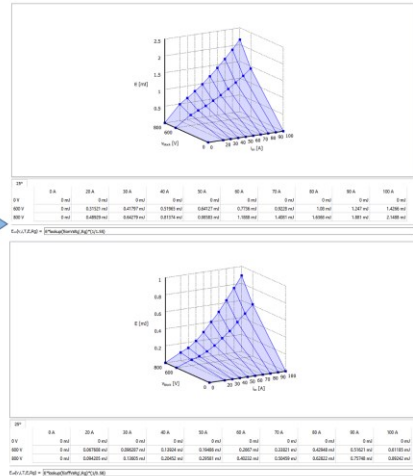
Figure 23. Clamped Inductive Switching Energy vs. Drain Current ( $V_{ds} = 600V$ )

Turn-on energy vs current

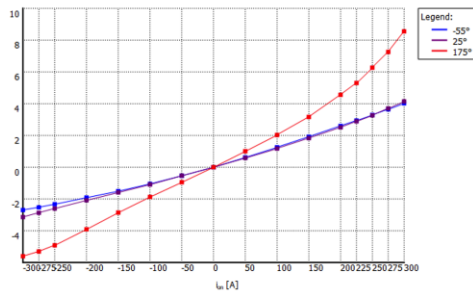
Extract from data sheet

Scalable for different gate resistance

Turn-off energy vs current



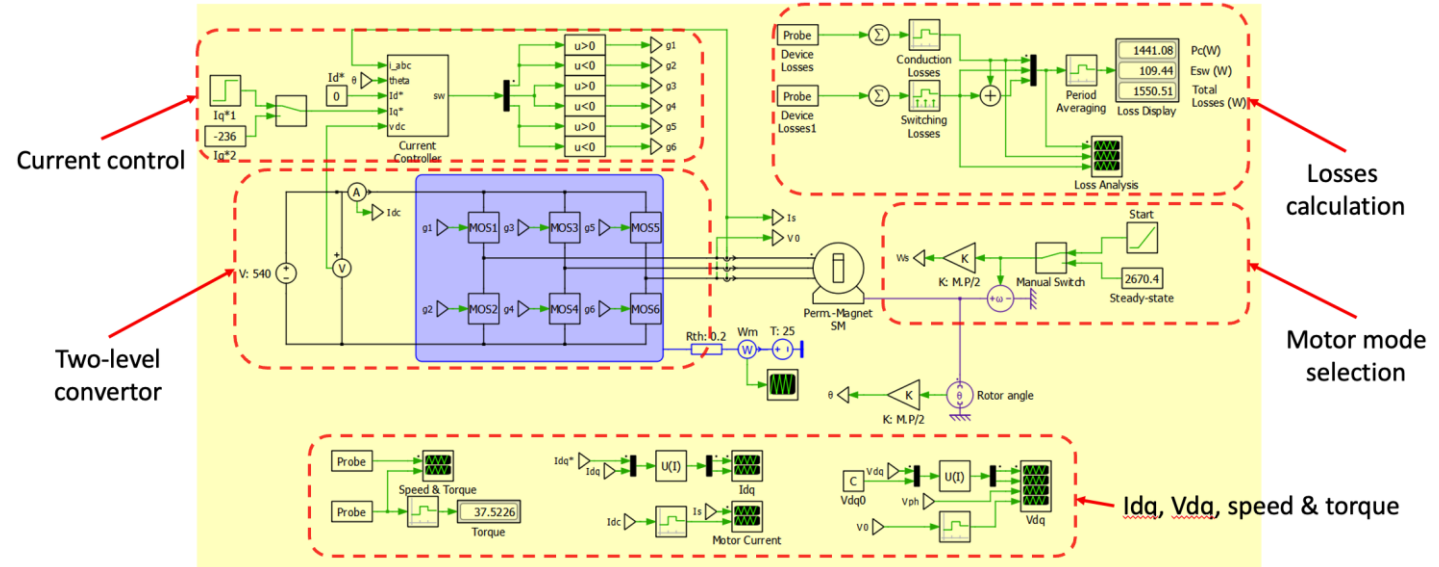
Output characteristics (First and third quadrant)



Thermal impedance

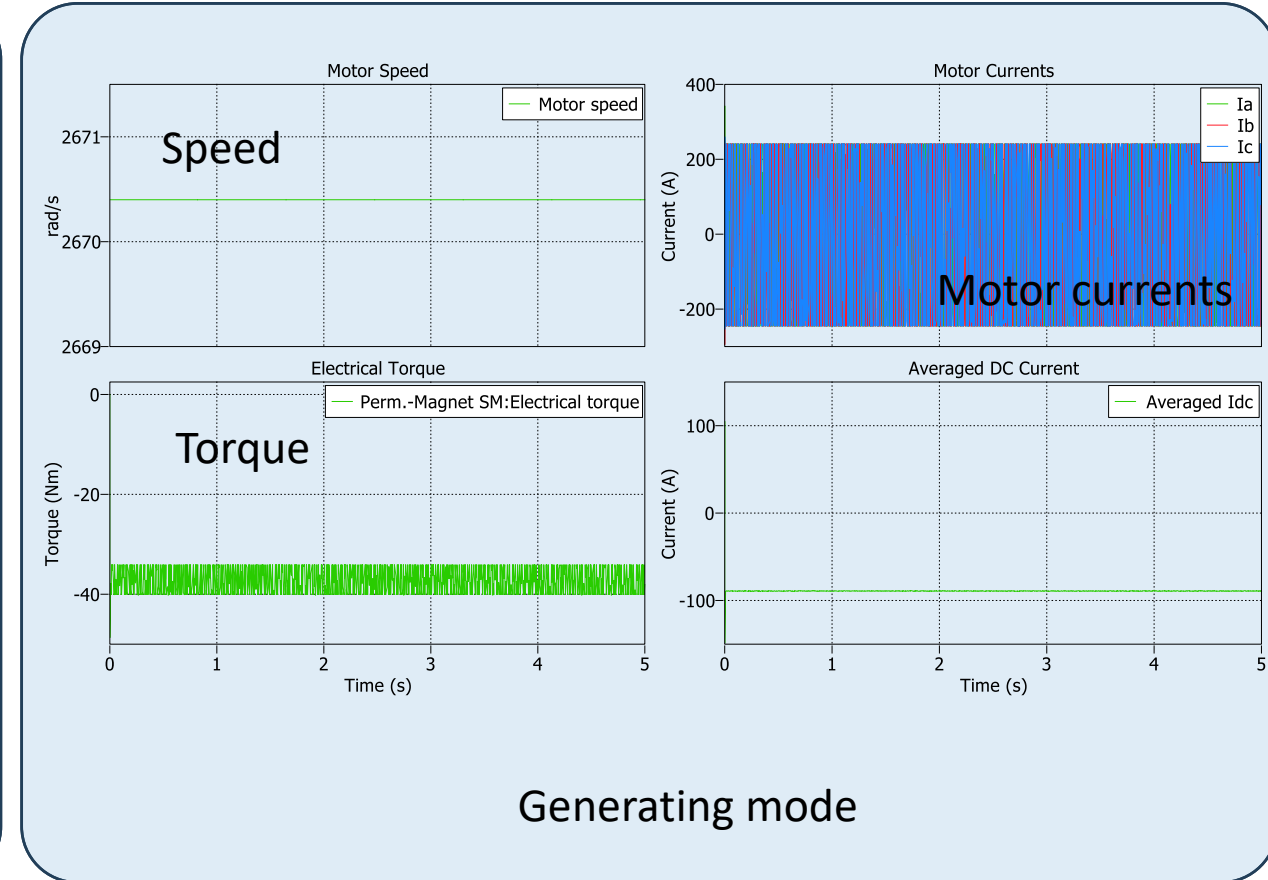
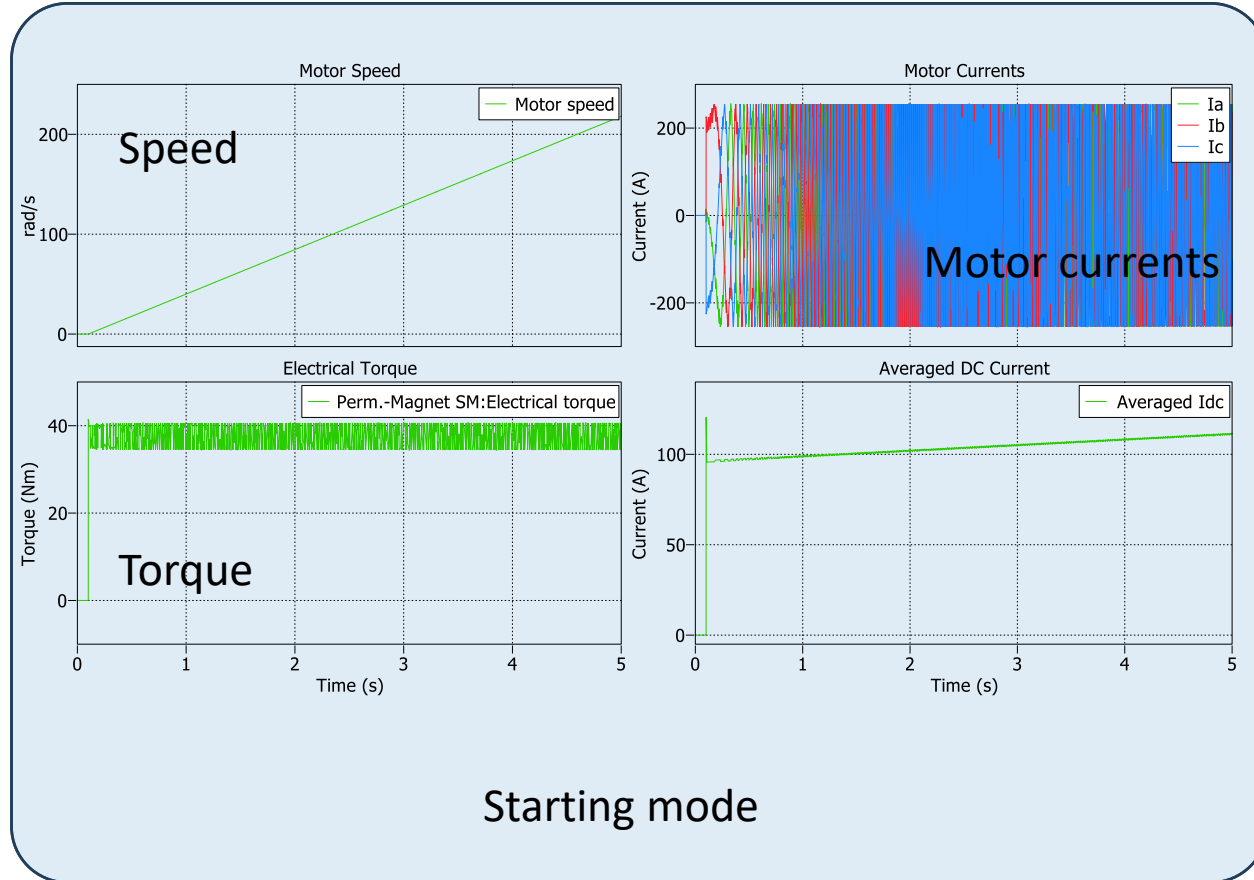
Type	1	2	3	4	5	6
R	0.0002146 K/W	0.001874 K/W	0.006369 K/W	0.02297 K/W	0.08657 K/W	0.1522 K/W
τ	1.066e-07 s	5.651e-07 s	1.107e-05 s	0.0001877 s	0.001925 s	0.02125 s

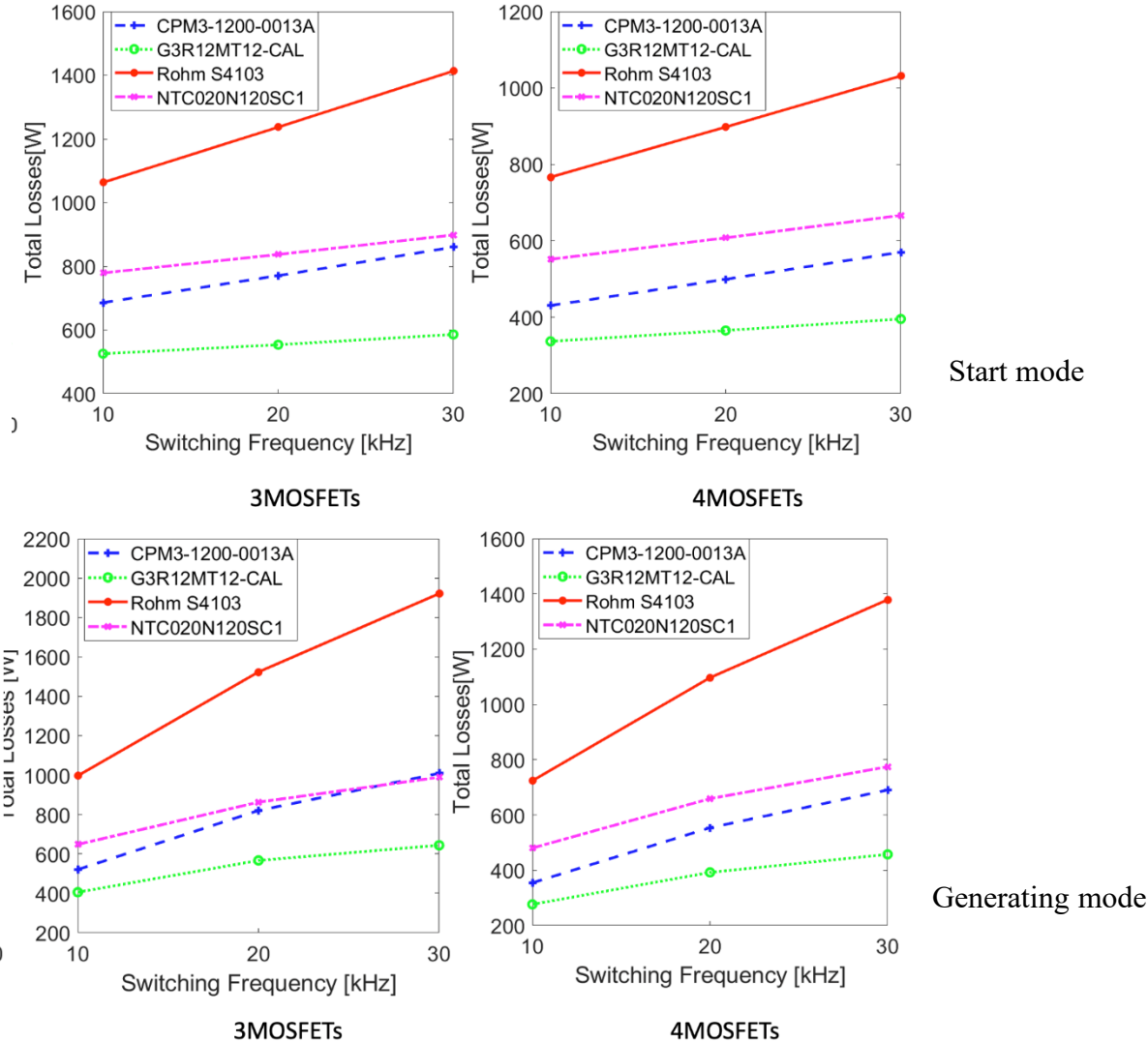
Describe its physical structure in terms of thermal transitions from the junction to case.



## Converter for Starter-generator

Power factor	$\cos\phi$	0-1
Load frequency	$f_{load}$	1kHz
Amplitude modulation ratio	m	0-1
Output power	$P_{out}$	100kW / 300kW
Output phase current	$I_{load}$	166.7Arms / 282Arms
Vdc		$\pm 270$ $\pm 540$

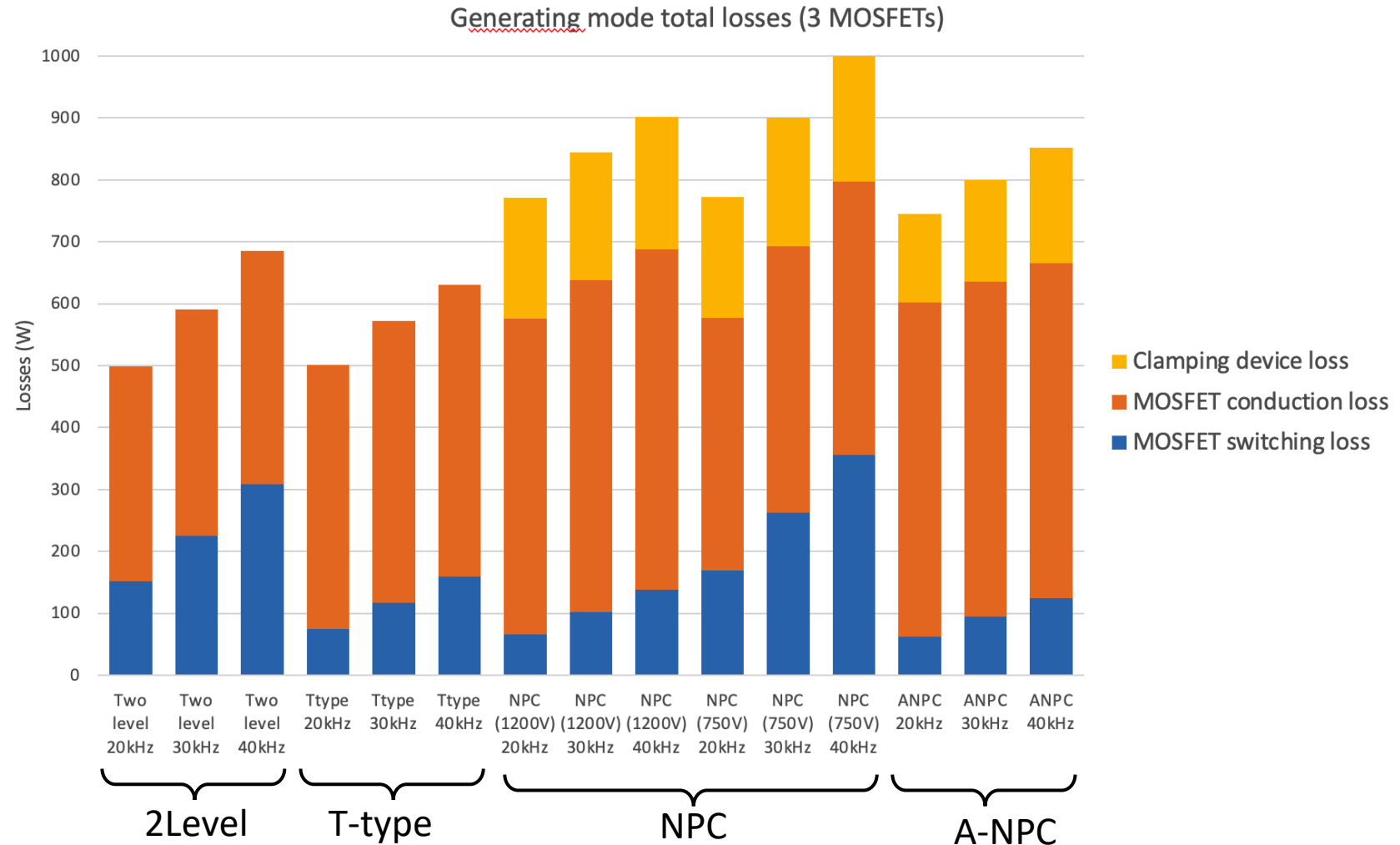




Parameter		Value
SiC MOSFET die	Wolfspeed Rohm GeneSiC On-Semiconductor	CPM3-1200-0013A S4103 G3R12MT12-CAL NTC020N120SC1
Switching frequency	$f_{sw}$	10kHz 20kHz 30kHz
Power factor	$\cos\phi$	0-1
Load frequency	$f_{load}$	1kHz
Amplitude modulation ratio	m	0-1
Output power	$P_{out}$	100kW
Output phase current	$I_{load}$	166.7A(RMS)

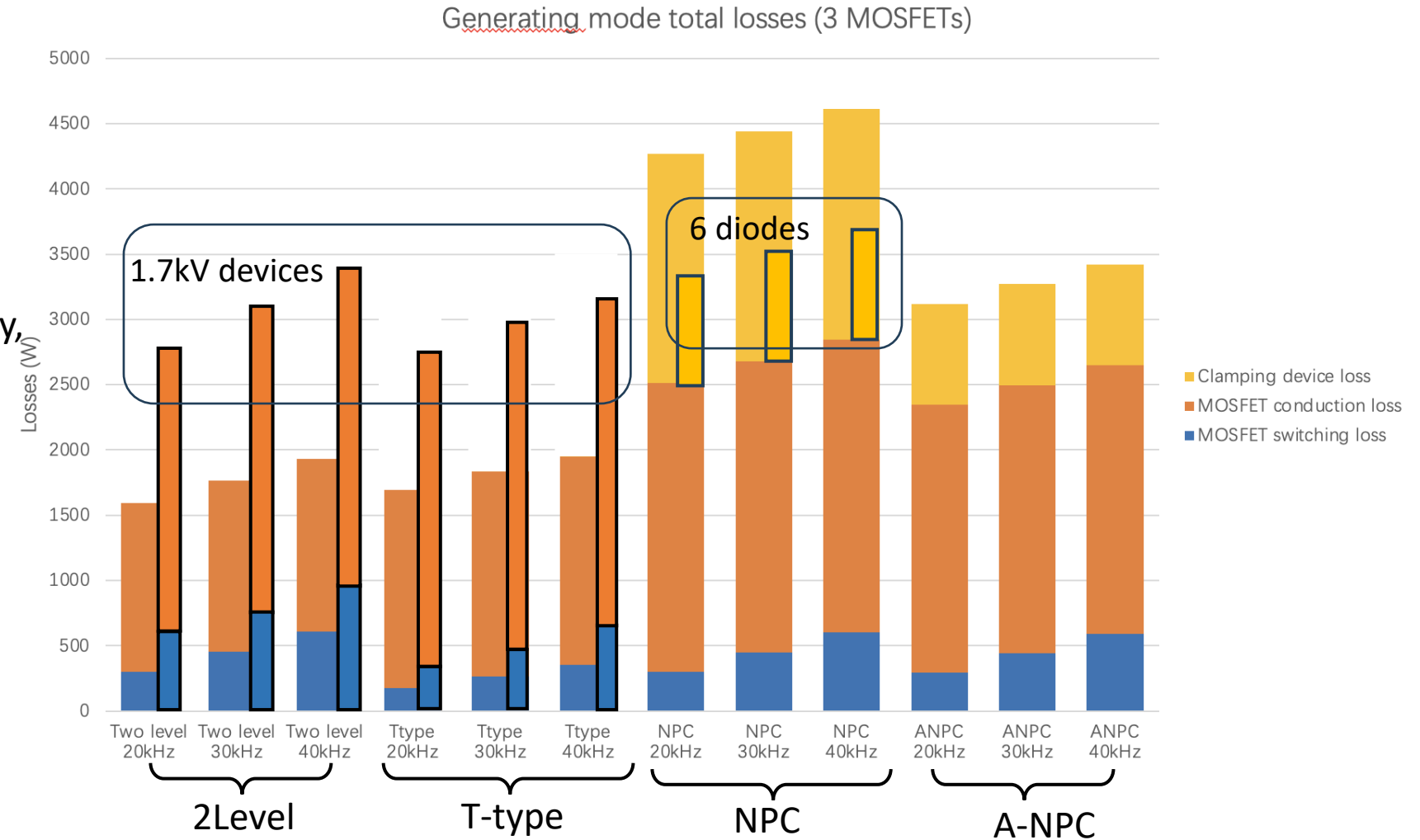
$$V_{DC} = 540V$$

- Two-level converter has the fastest increase of losses as the switching frequency increases
- T-type has the overall highest efficiency thanks to lower switching losses (lower switching voltages)
- NPC penalised by higher conduction losses
- Availability of low  $R_{DS,on}$  devices at <1.2kV



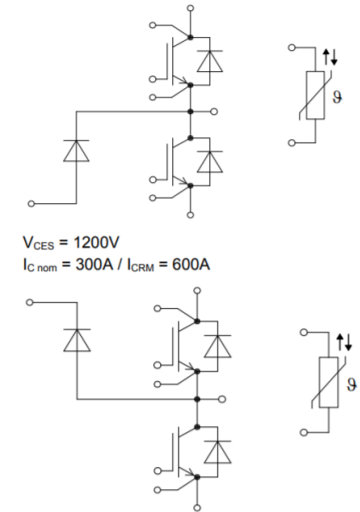
1080V 282A rms

- 1.2kV cannot be used due to required  $\approx 2x$  voltage margin
- 1.7kV increase losses considerably, making NPC competitive for



# Power modules for MEA

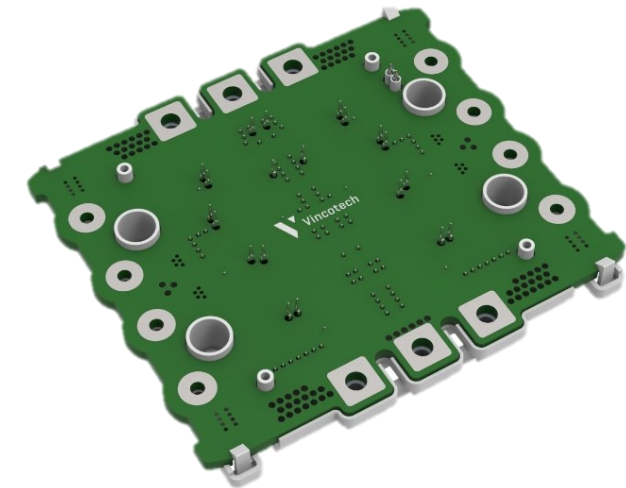
- There is no SiC multi-level module on the market beyond small (<50A) prototypes/samples
- Similarly rated Si IGBT modules have equivalent on-state resistance double of what can be achieved with SiC ( $5.7m\Omega$  vs  $2.6m\Omega$ )
- MuSiCA (Multilevel SiC power modules for Aerospace applications) to enable future converters in MEA with DC voltage above 1000V



- Modularity
- MRL



- MuSiCA will enable converters at 20kW/kg compared with 10kW/kg of the Si-IGBT state of the art

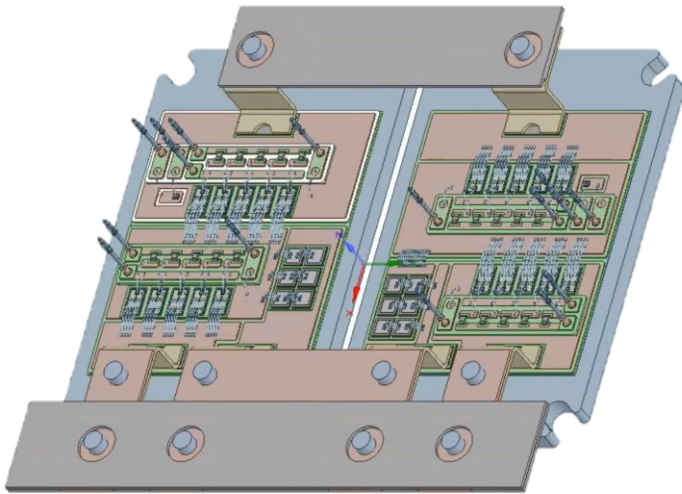


Specifications		Notes
Topology	3-Level NPC	3-Level I-type NPC Converter
DC-link voltage	540V/1080V	1080V ( $\pm 540V$ ) max DC-link voltage
Power	100kVA per phase	power factor between 1 and 0
Current	280A/400A	Steady State current >280A(rms) Peak AC current of 400A 400A(maximum) peak of AC current under highest load condition.
Switching frequency	>10kHz	No maximum
Thermal monitoring	required	NTC/PTC
Passives		DC-link capacitor.
Cooling method	Air cooling	The modules are mounted on an Aluminium heat sink. Assume base plate temperature up to 80°C, worst case scenario. Air 'coolant' temperature according to airplane requirements. Max air temp. 50°C.

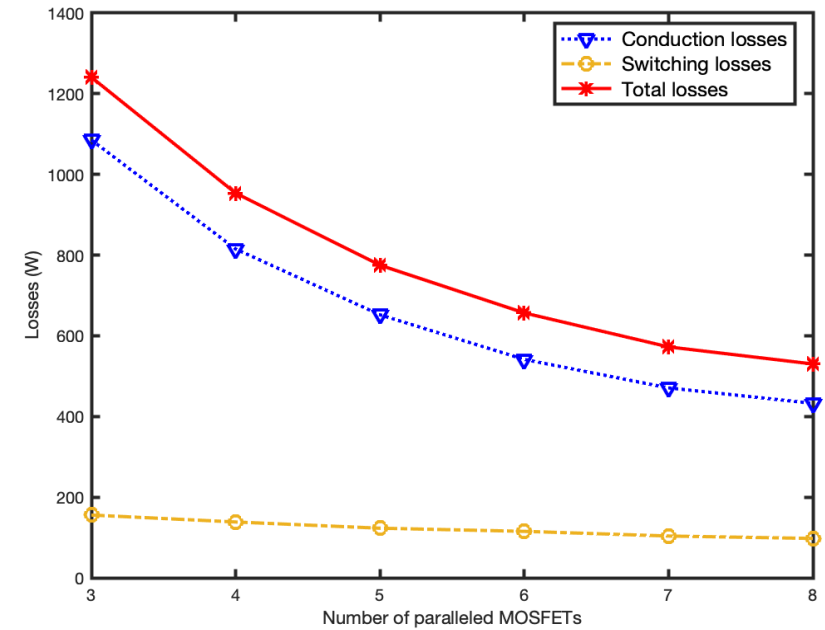


## Electro-thermal simulations

- Scalable design for different power levels/applications
- 5 dies per switch → >99% efficiency
- 6 SiC diodes

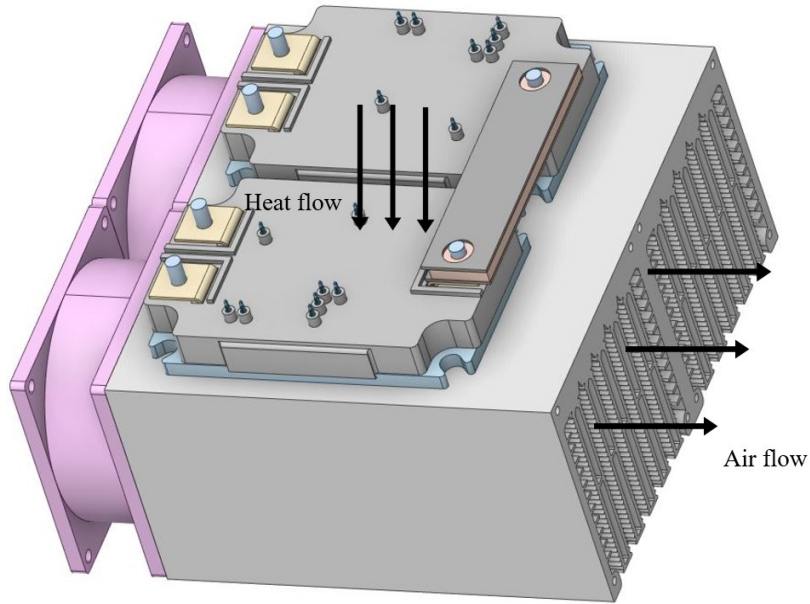


*Losses for different number of MOSFETs in parallel (Devices characteristic at Temp.=125 °C)*



Losses per phase Leg [W]	Conduction losses	Switching losses	Total losses	Efficiency
3 MOSFETs per Switch	1085.4	155.6	1241.0	98.74%
4 MOSFETs per Switch	815.1	138.4	953.5	98.98%
5 MOSFETs per Switch	652.1	123.0	775.1	99.21%
6 MOSFETs per Switch	541.4	115.2	656.6	99.33%

# Power modules for MEA: Electro-thermal optimisation



3D view of forced air-cooled system

## Three MOSFETs per switch

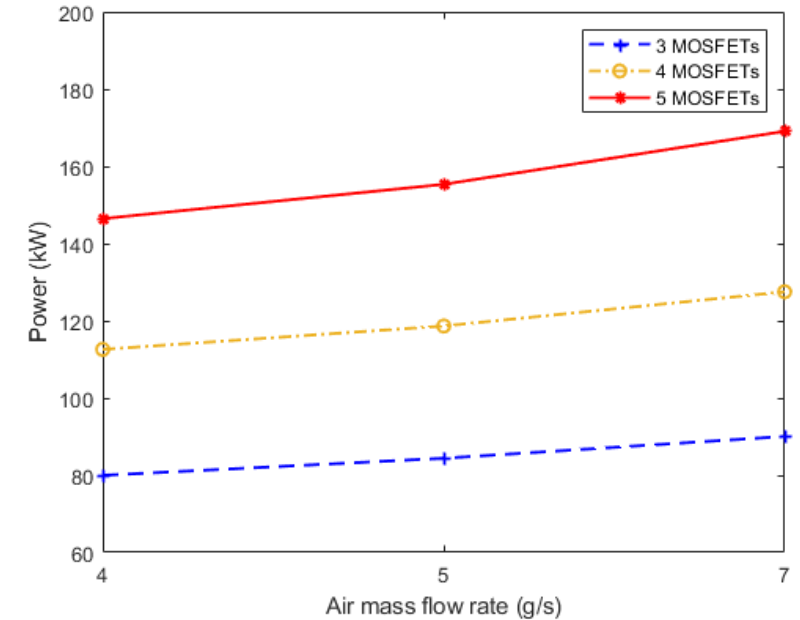
Temperature (°C)	4g/s	5g/s
Top MOSFETs	197.4	188.9
Bottom MOSFETs	175.4	166.8
Middle MOSFETs	223.5	213.1
Clamping diodes	150.3	141.9

## Four MOSFETs per switch

Temperature (°C)	4g/s	5g/s
Top MOSFETs	152.7	146.1
Bottom MOSFETs	139.5	132.5
Middle MOSFETs	166.0	158.8
Clamping diodes	142.4	135.1

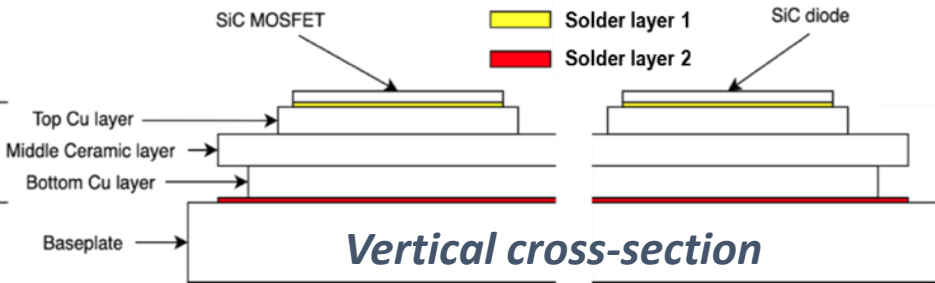
## Five MOSFETs per switch

Temperature (°C)	4g/s	5g/s	7g/s
Top MOSFETs	124.4	118.4	110.6
Bottom MOSFETs	115.9	110.0	101.9
Middle MOSFETs	133.0	126.8	118.9
Clamping diodes	133.4	127.2	118.7

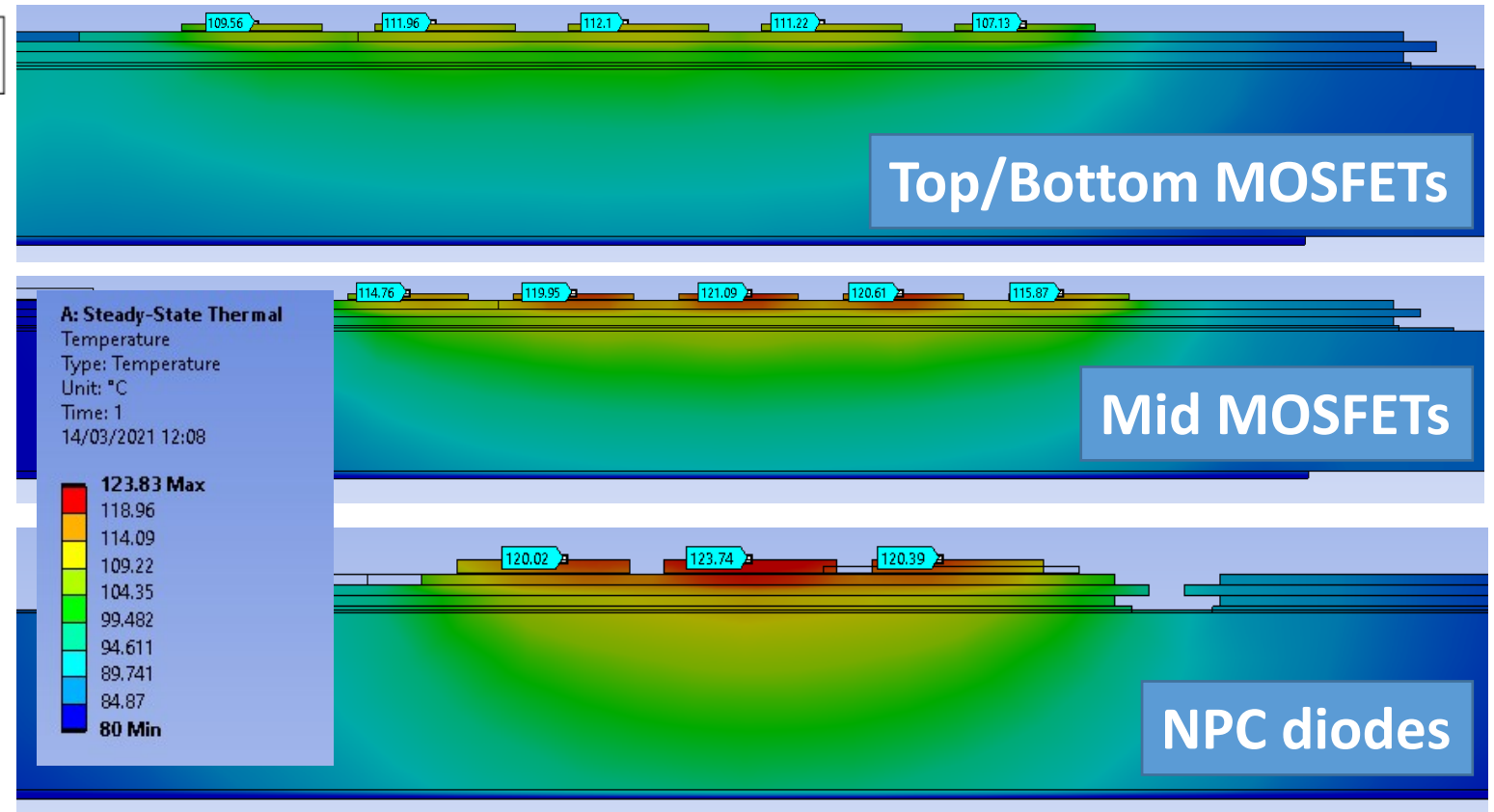
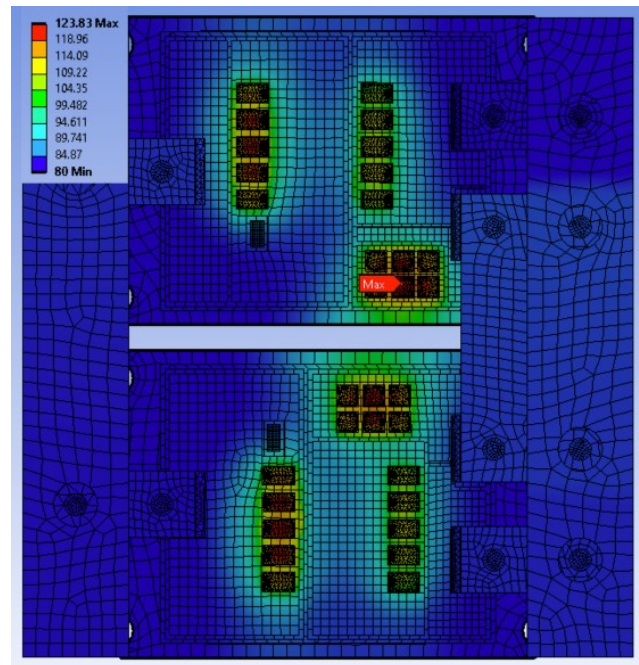


Achievable power at different flow rates

# Power modules for MEA: Electro-thermal optimisation

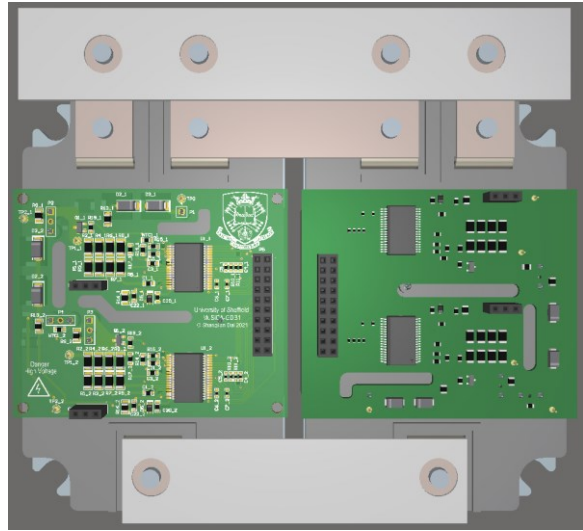


Max temperature increase at full load = 44°C

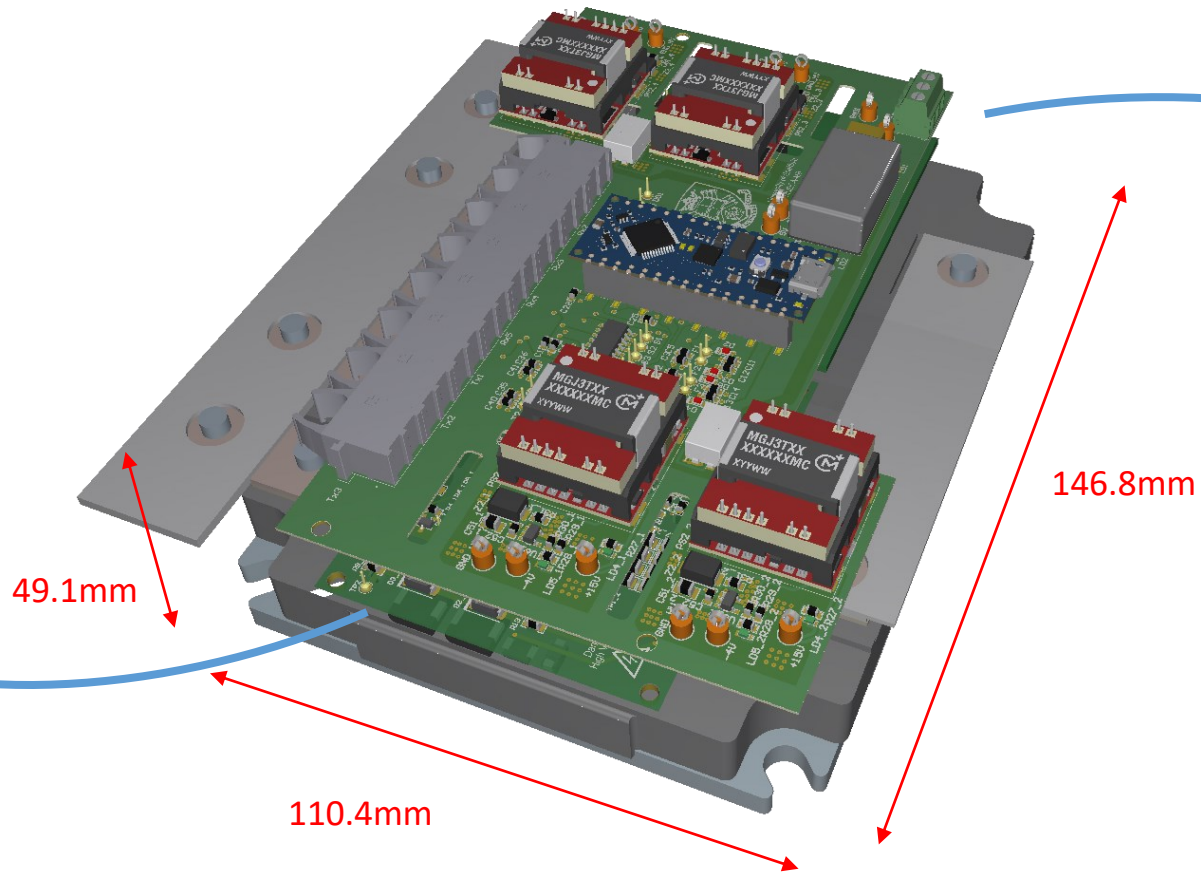


*temperature distribution*

# Power modules for MEA: Gate drives

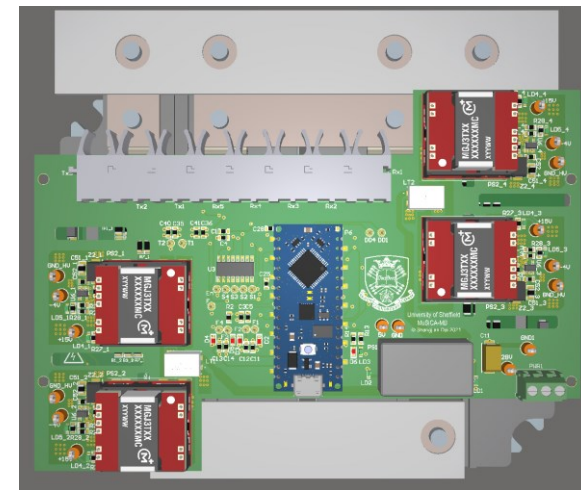


Sub-gate drive boards



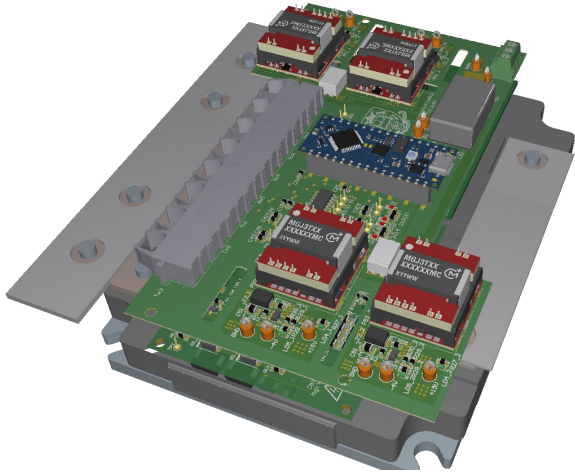
Total cuboid volume=0.8L

Master board

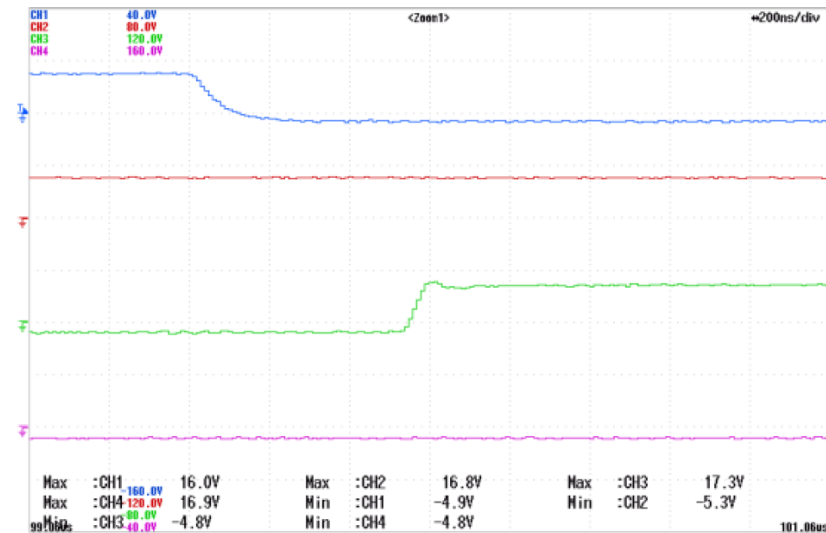
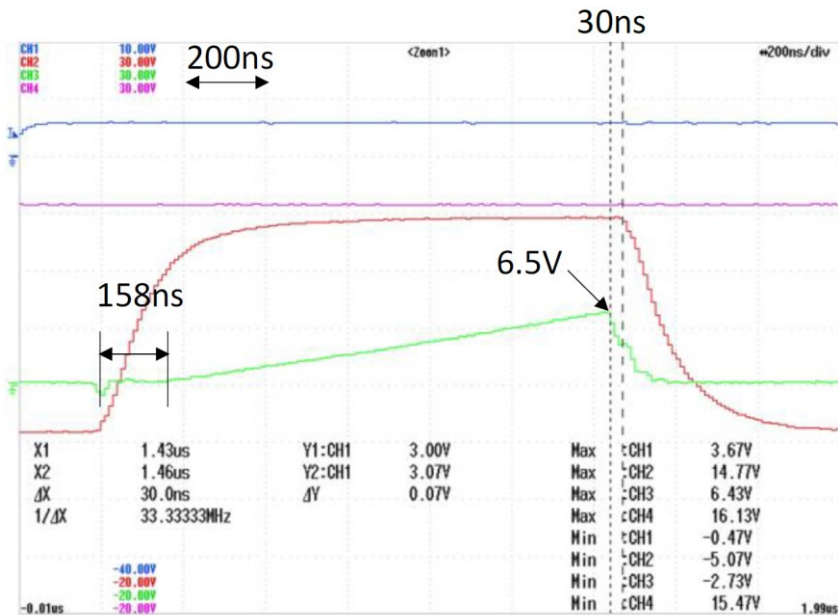




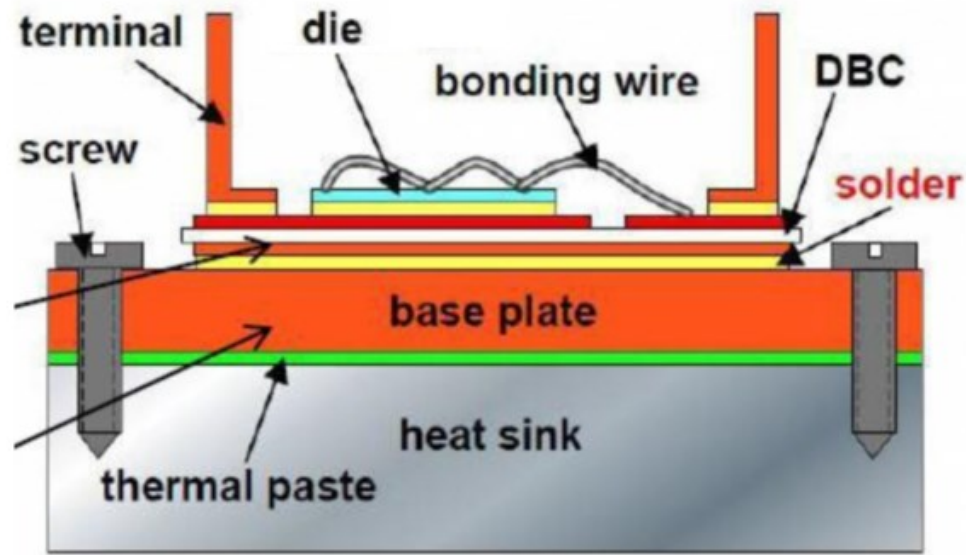
# Power modules for MEA: Gate drives



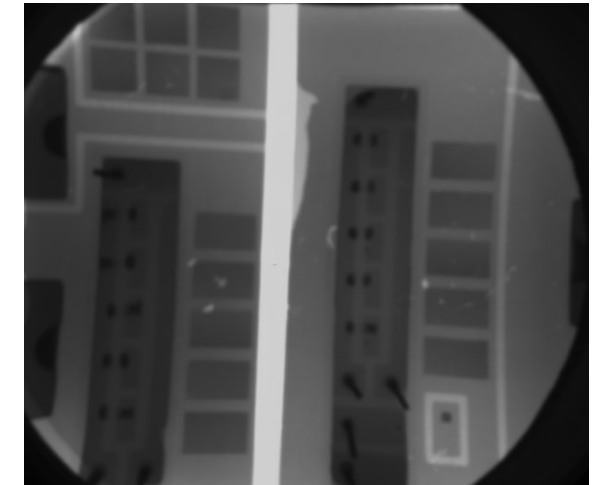
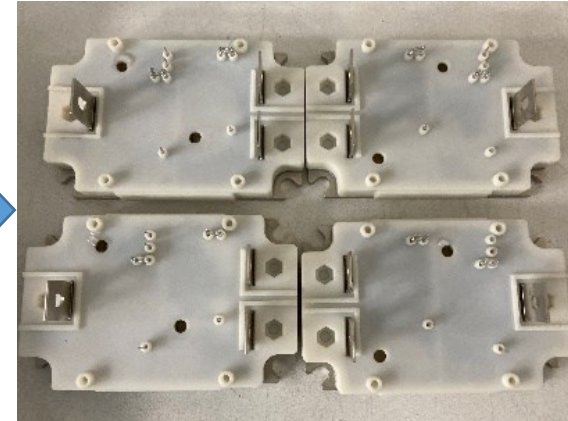
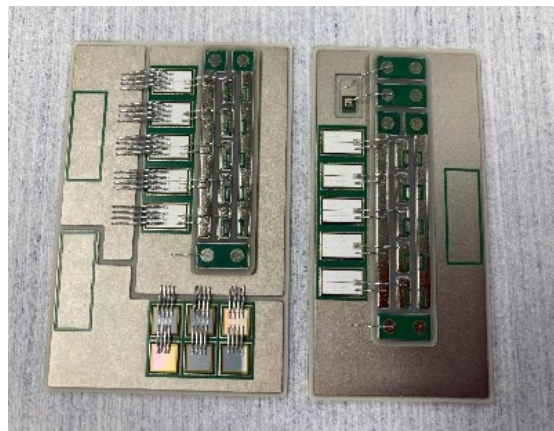
- Programmable gate drive
- $-4, +15V$
- 30A peak at 50kHz
- Programmable short circuit protection, desaturation detection  $V_{ds,th} = 6.5V$  ( $I_{sc} = 1346A$ ); 400 – 1000ns
- Hardware deadtime 105ns – 4445ns with the step of 70ns



# Power modules for MEA: packaging



- Gen3 SiC MOSFET from Wolfspeed
- Si<sub>3</sub>N<sub>4</sub> active metal brazed (AMB) substrate
- AlSiC baseplate (air cooling)
- High temperature grade silicon gel (-50 to 150 °C)
- Vacuum solder reflowing for die attach & sub attach
- Wire bonding & soldered terminals

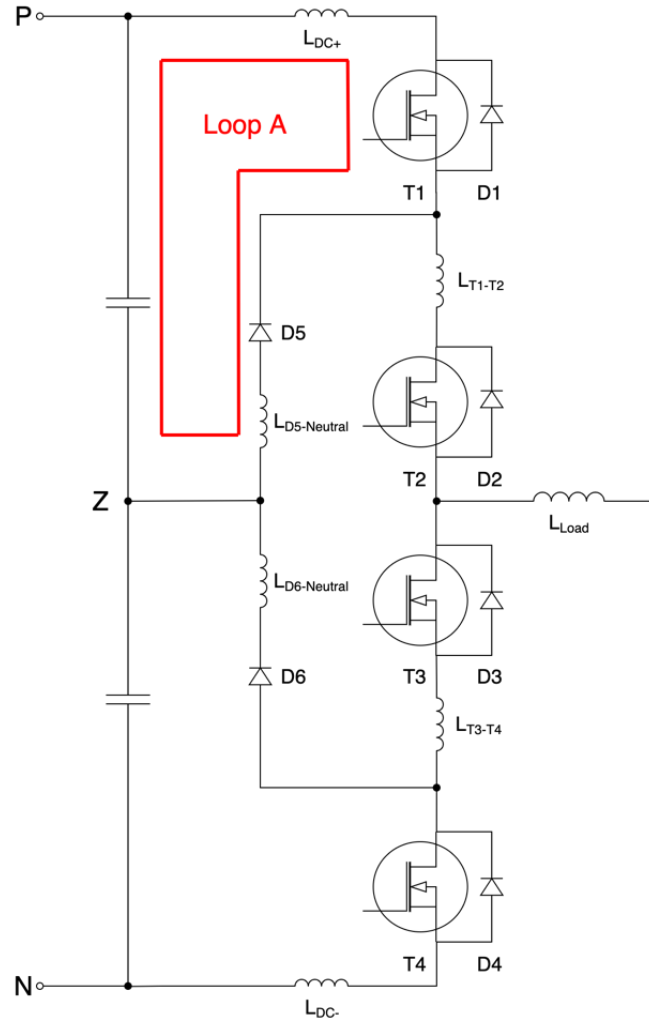


Die attach & wire bonding

Substrate attach & terminal bonding

Housing & encapsulation

X-ray transmission image

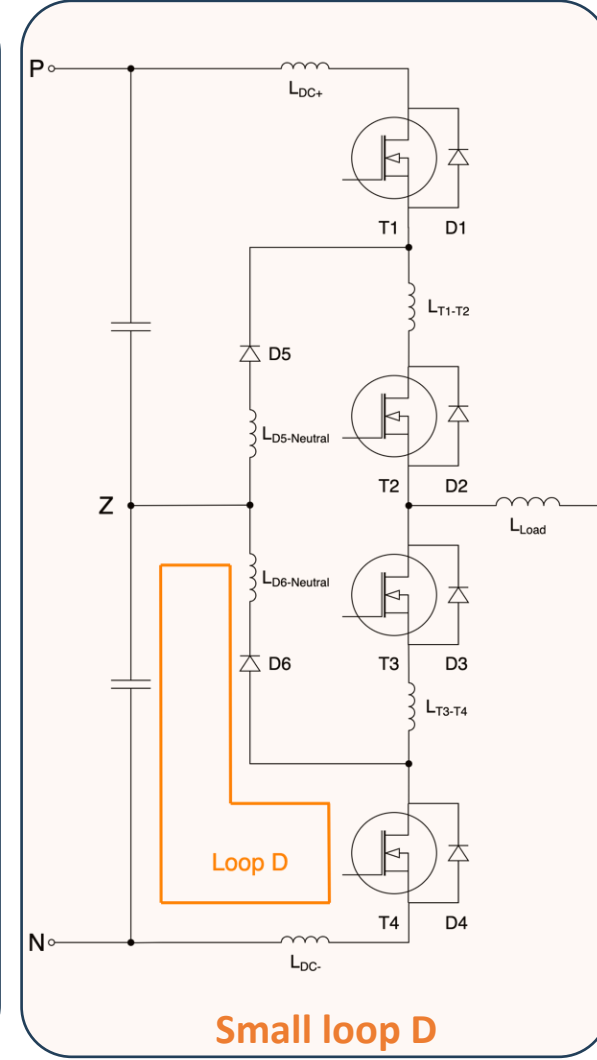
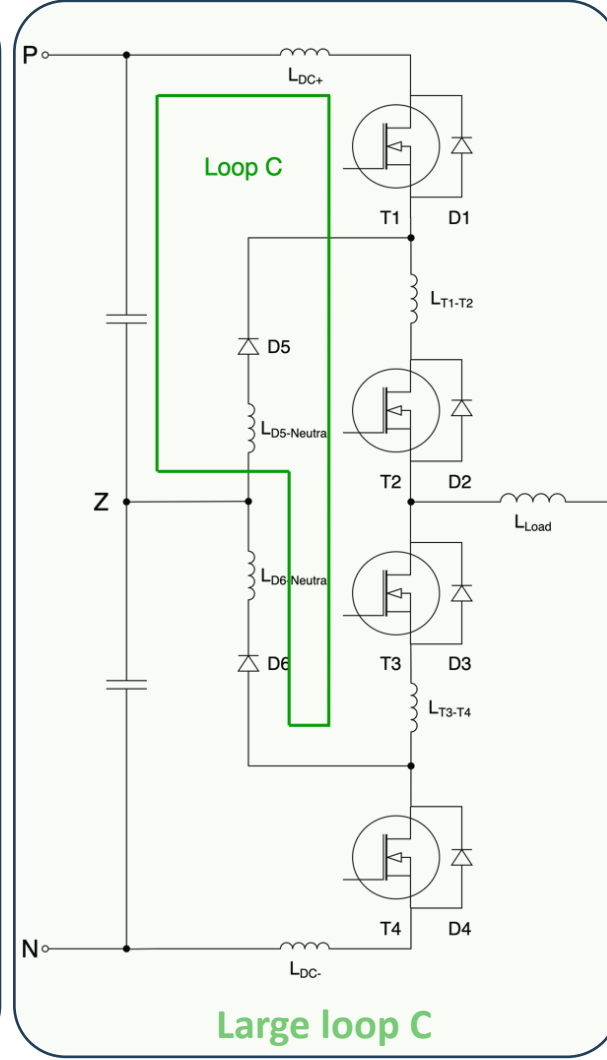
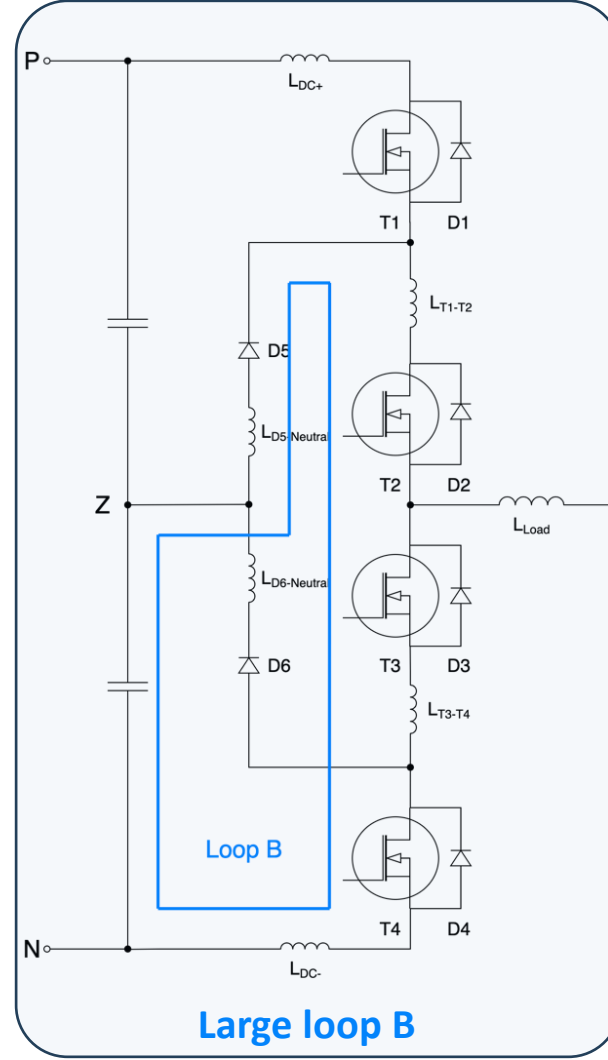
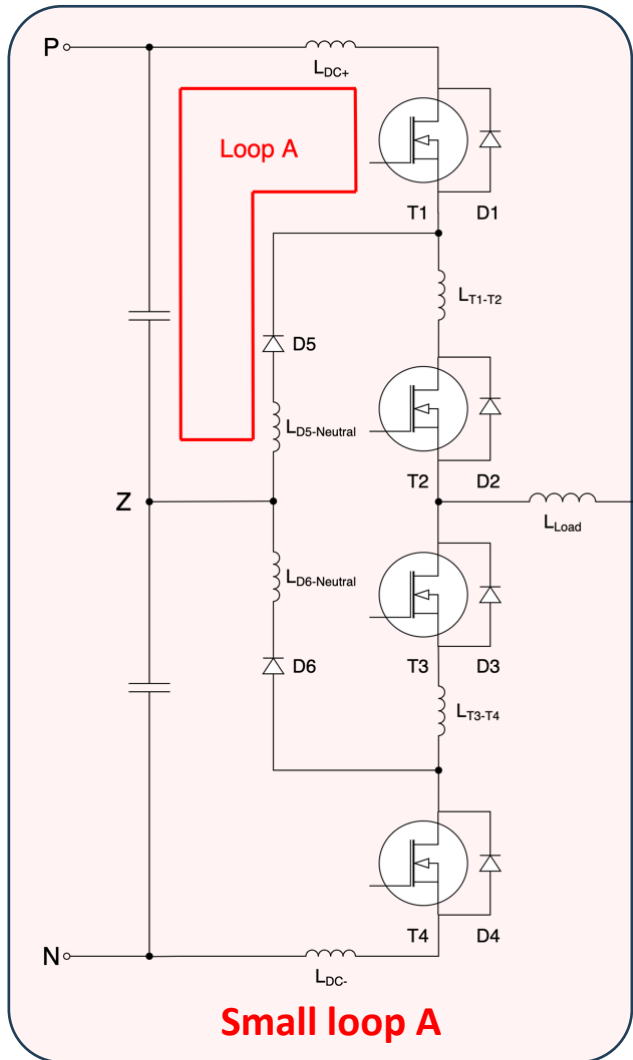


Fast-switching transients of SiC devices result in high current gradients  $di/dt$ , which can be over an order of magnitude higher than those of conventional Si devices.

Fast current transients during switch commutation result in over-voltages  $L \cdot di/dt$  due to parasitic inductances

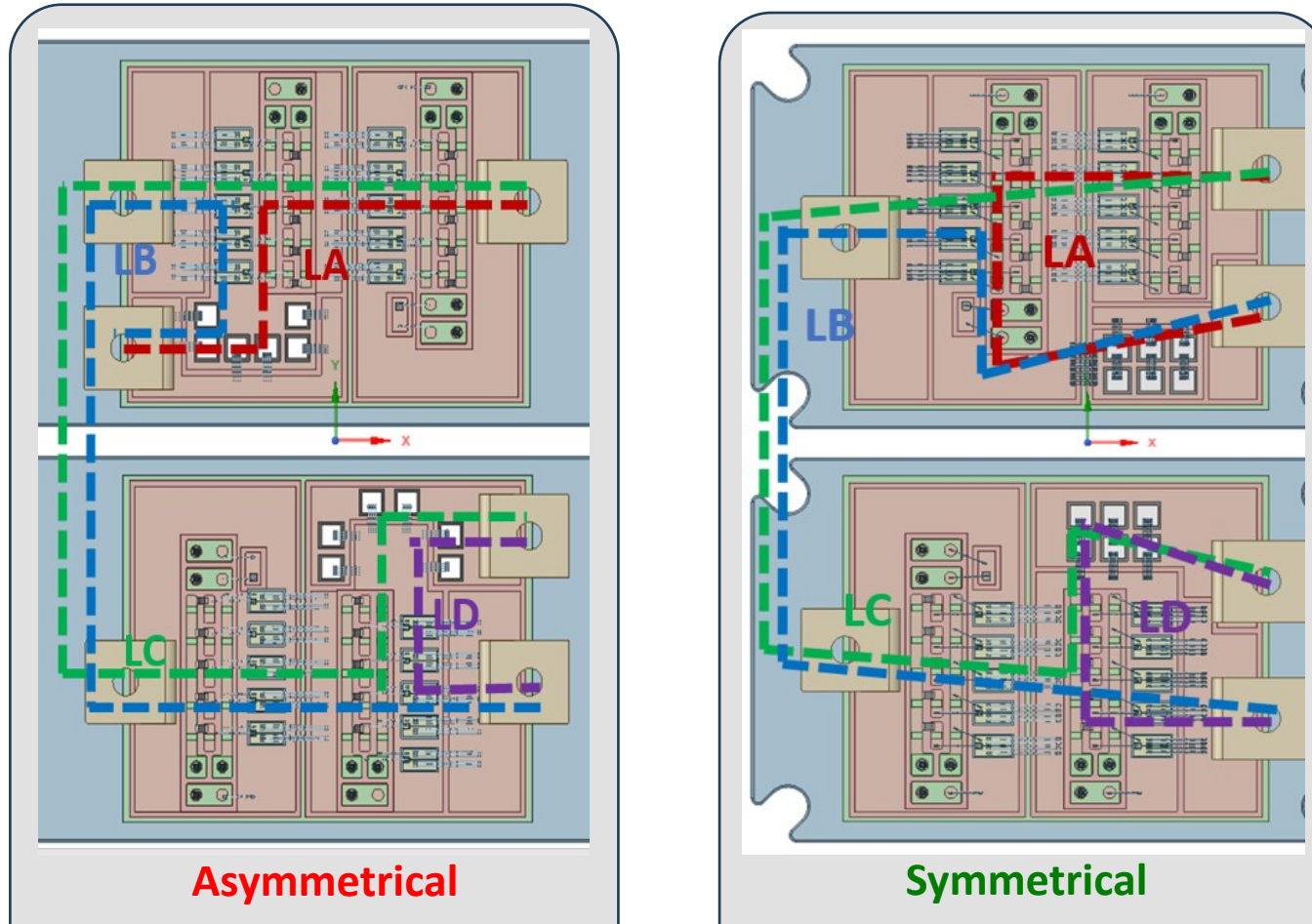
- electrical stress and potential reduction in the safe operating area of devices,
- reduced reliability
- increased electromagnetic interference (EMI) emissions

# Power modules for MEA: parasitics





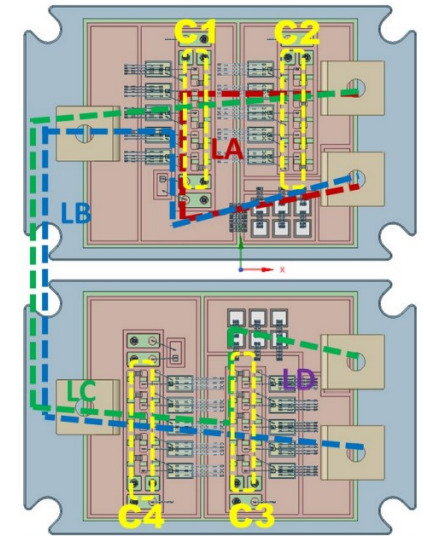
# Power modules for MEA: parasitics



**Asymmetrical**

**Symmetrical**

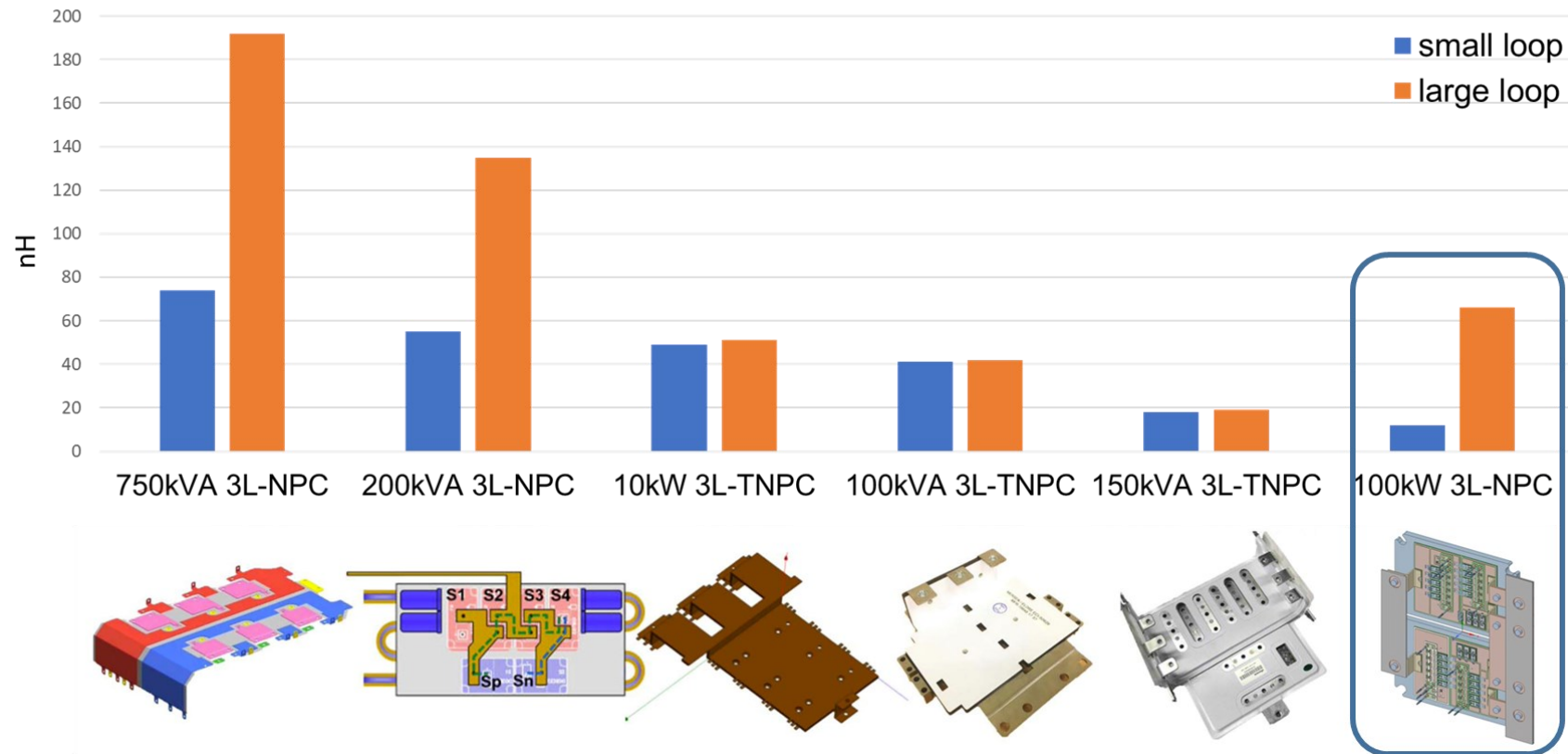
Design	Loop A	Loop B	Loop C	Loop D
Asymmetrical	<b>22.2nH</b>	68.5nH	76.9nH	<b>9.3nH</b>
Symmetrical	<b>11.1nH</b>	75.3nH	74.6nH	<b>9.9nH</b>



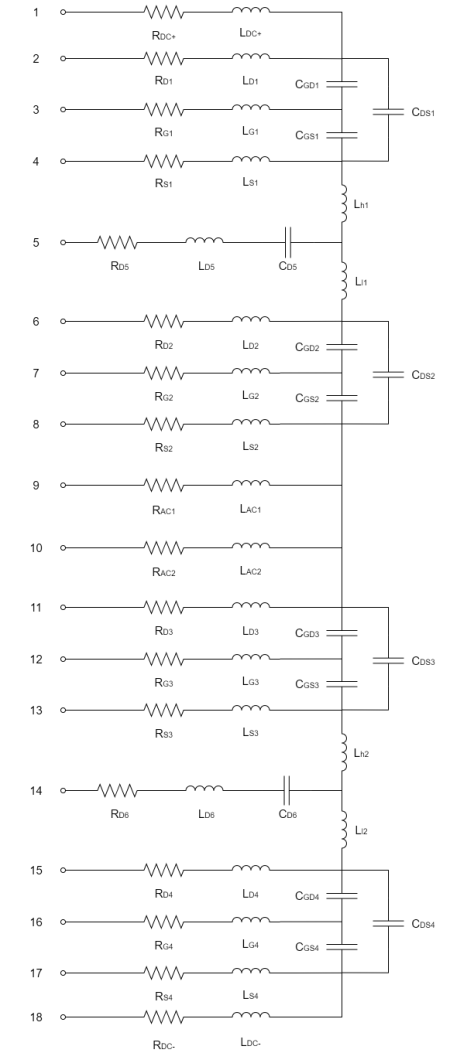
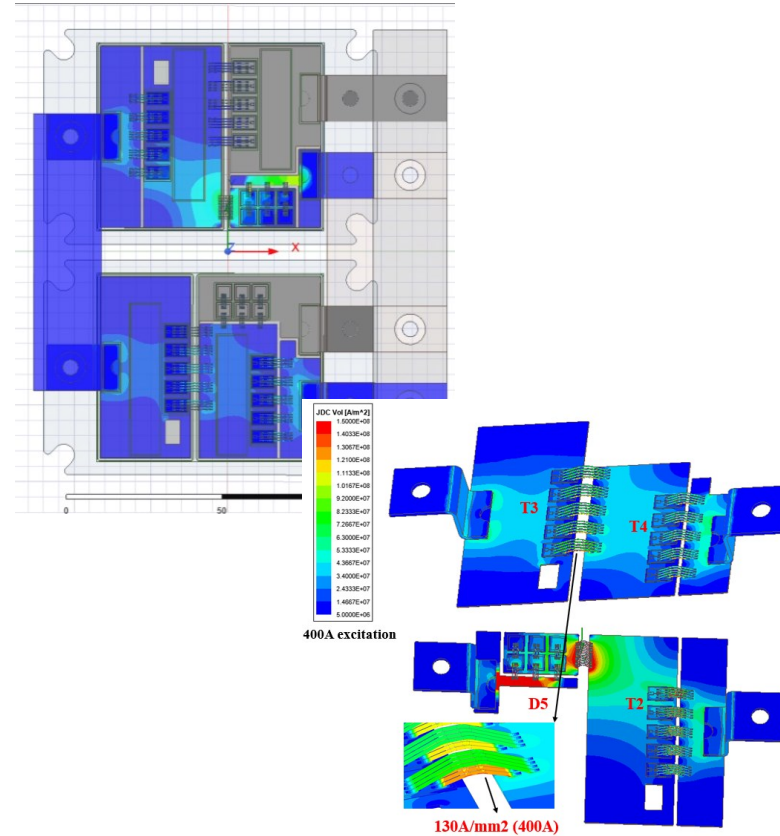
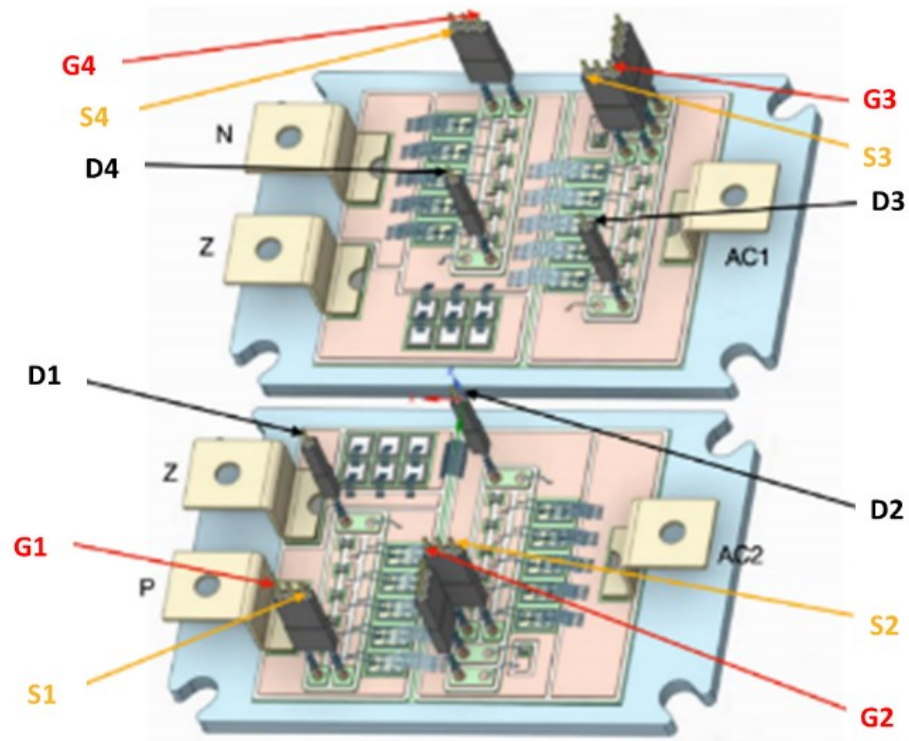
Minimization of coupling between current and control loops

Control Loops	Self-inductance	Mutual inductance	
		LA	LB
C1	17.09nH	0.16	0.93
C2	16.95nH	0.48	0.97
C3	17.15nH	0.03	0.70
C4	17.06nH	0.03	0.78

## Different power module loop inductance comparison



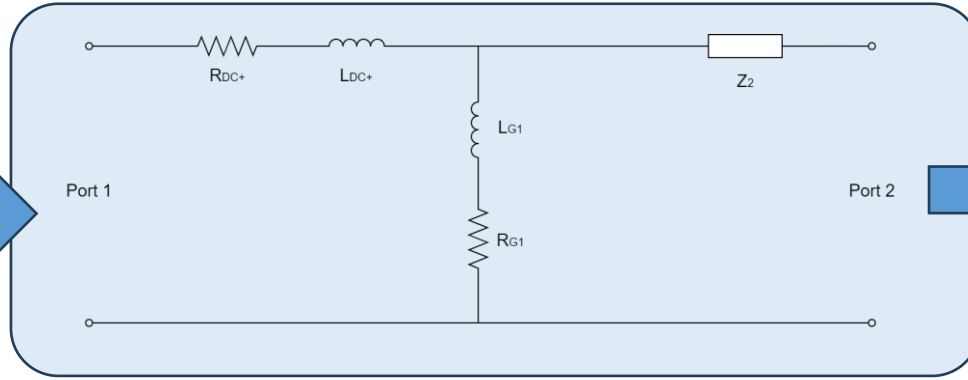
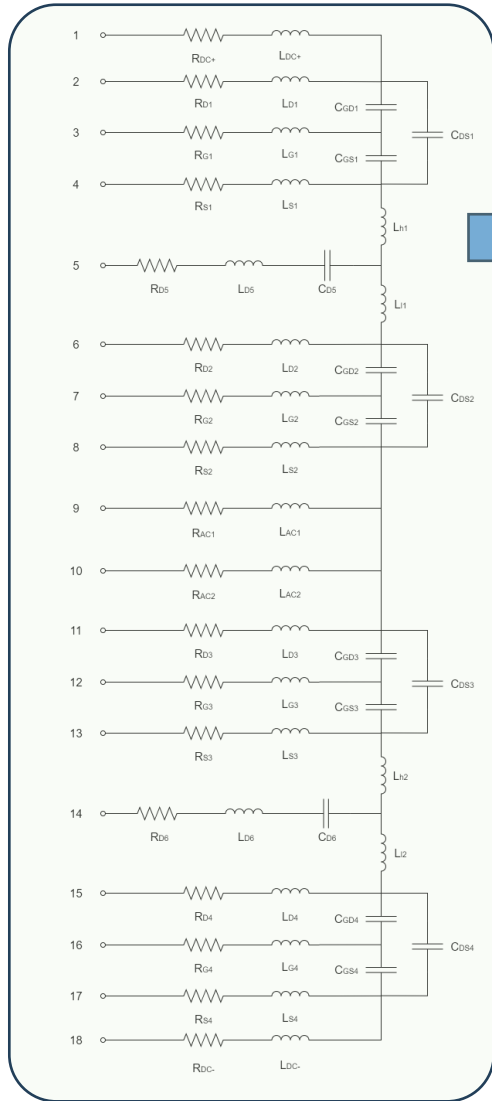
# Power modules for MEA: parasitics



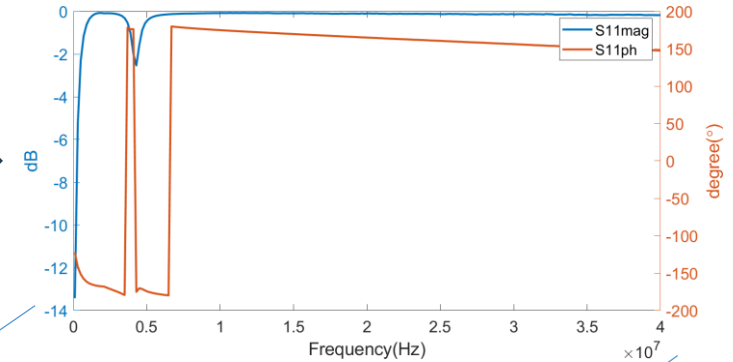
Power module commutation inductances can be calculated using analytical tools

- method of moments e.g. Ansys Q3D
- Direct measurement is difficult in complex structure e.g. multi-level modules and is not possible in assembled modules

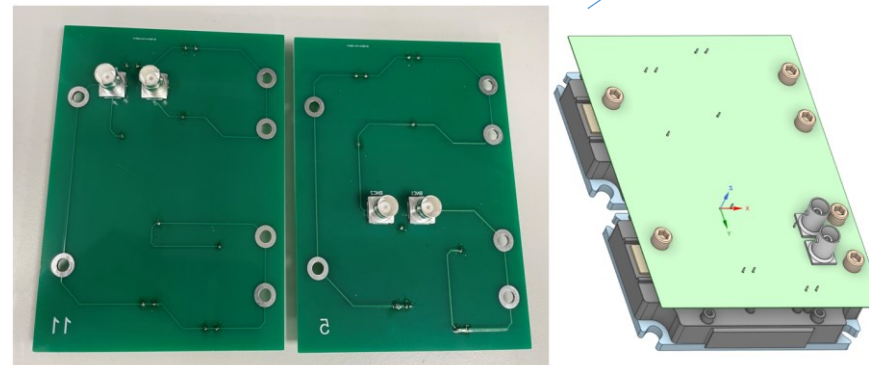
# Power modules for MEA: parasitics



to measure all the parasitic inductances of each branch, the small signal model can be subdivided in **16 two-port networks**



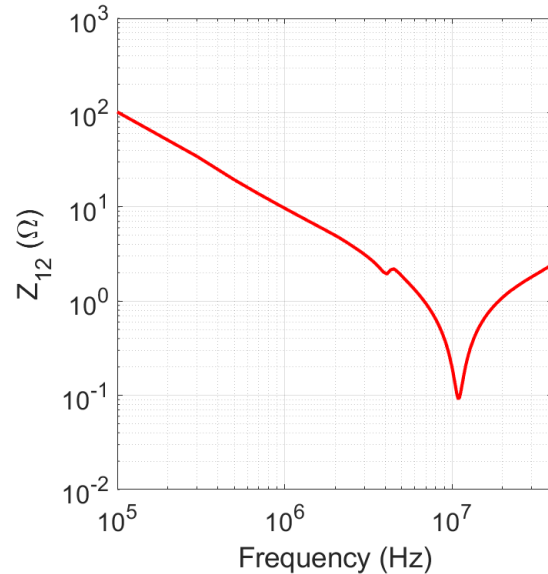
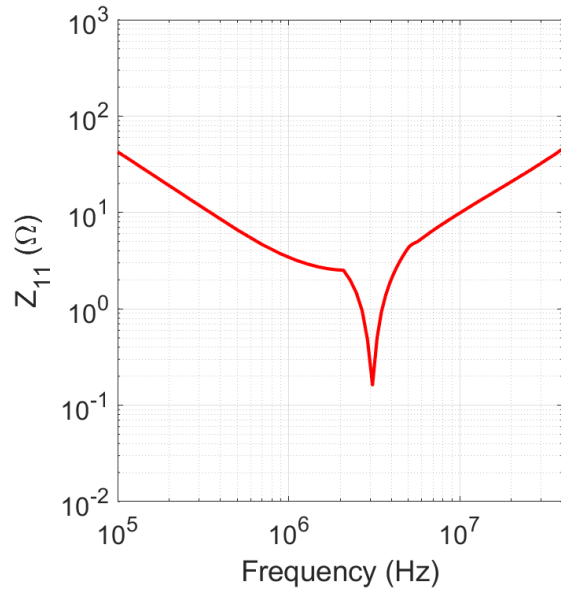
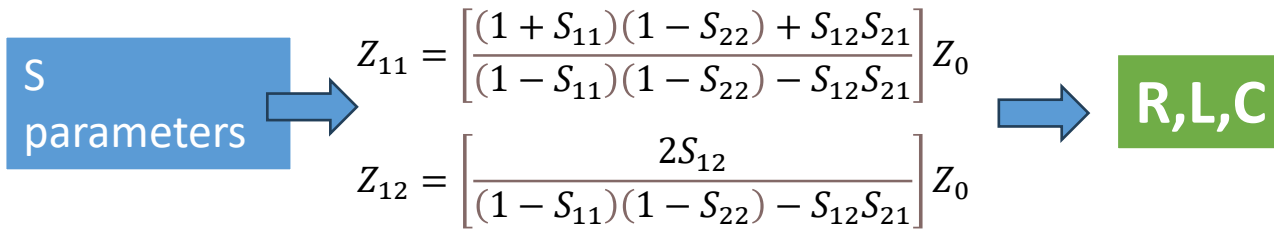
Measure S-parameters with VNA



- two-port 'Short-Open-Load-Thru' calibration to eliminate the effects caused by the cable
- decoupling PCBs for the measurement of the PCB parasitics

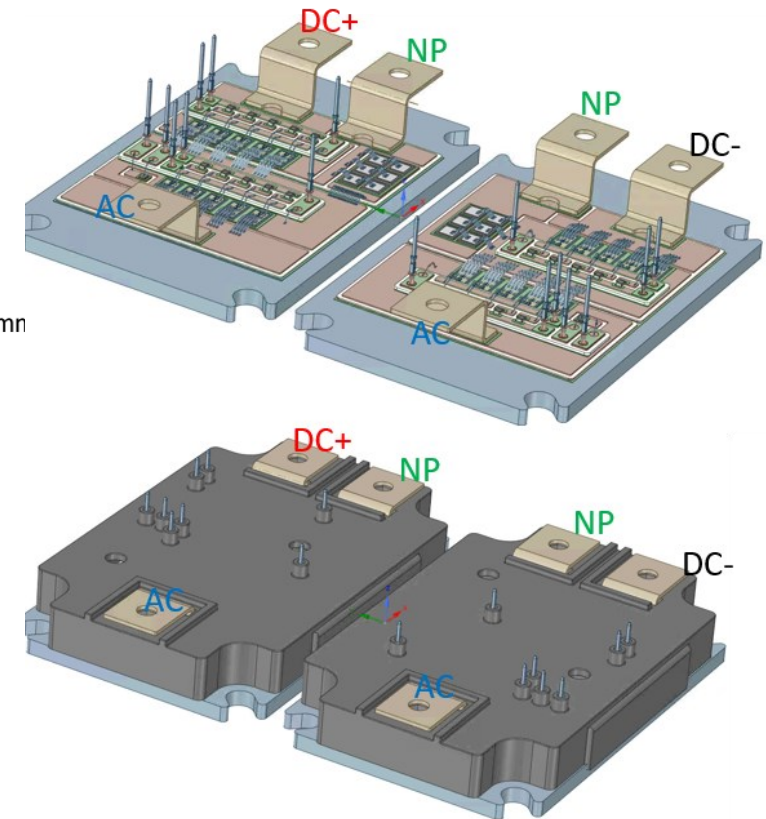
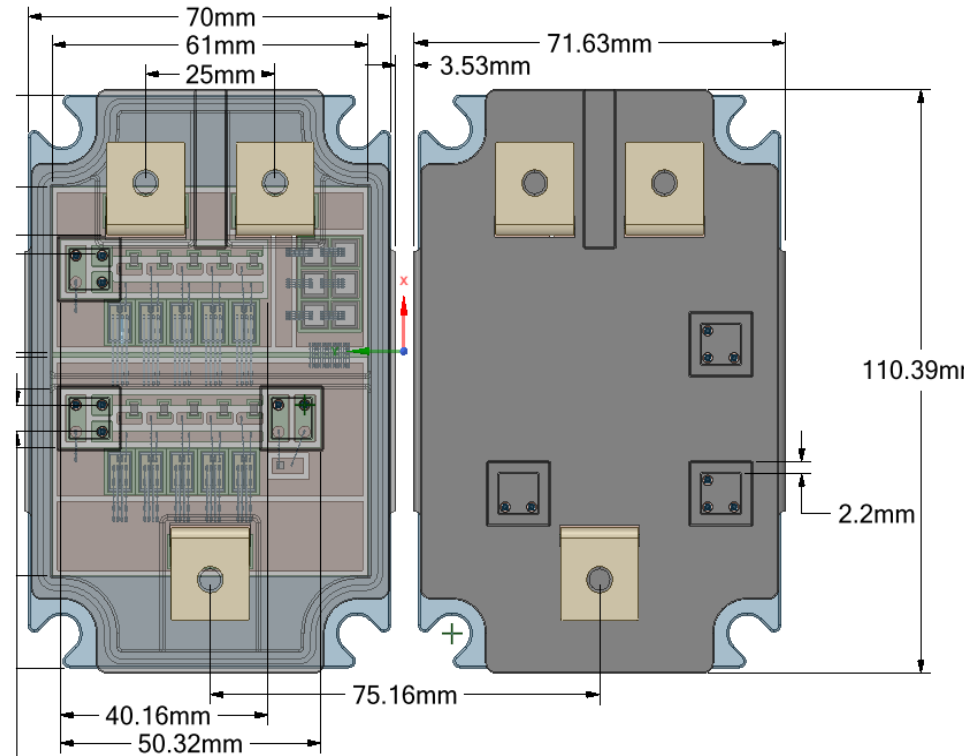
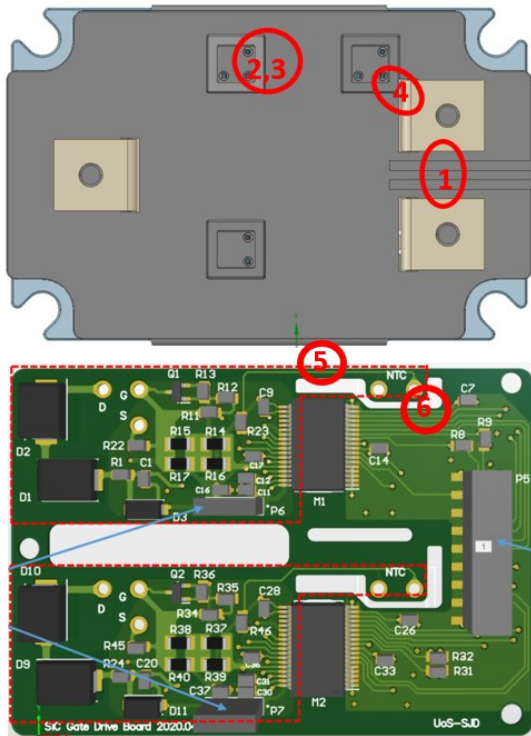


# Power modules for MEA: parasitics



Parasitic inductance	Simulation (nH)	Measurement (nH)	%difference
D1 (T1 MOSFET)	20.61	22.49	8.7%
G1 (T1 MOSFET)	22.64	20.97	7.6%
S1 (T1 MOSFET)	20.71	20.53	0.9%
D2 (T2 MOSFET)	20.74	20.78	0.2%
G2 (T2 MOSFET)	21.92	19.94	9.5%
S2 (T2 MOSFET)	20.72	21.28	2.7%
D3 (T3 MOSFET)	21.27	23.11	8.3%
G3 (T3 MOSFET)	22.64	23.31	2.9%
S3 (T3 MOSFET)	20.67	20.72	0.2%
D4 (T4 MOSFET)	20.62	21.88	5.9%
G4 (T4 MOSFET)	22.59	22.95	1.6%
S4 (T4 MOSFET)	20.72	21.08	1.7%
DC+	9.58	10.16	5.9%
DC-	7.75	8.25	6.3%
AC1	7.72	8.46	9.1%
AC2	9.11	9.74	6.7%
D5 (Clamping diode)	11.82	10.99	7.3%
D6 (Clamping diode)	11.89	12.87	7.9%

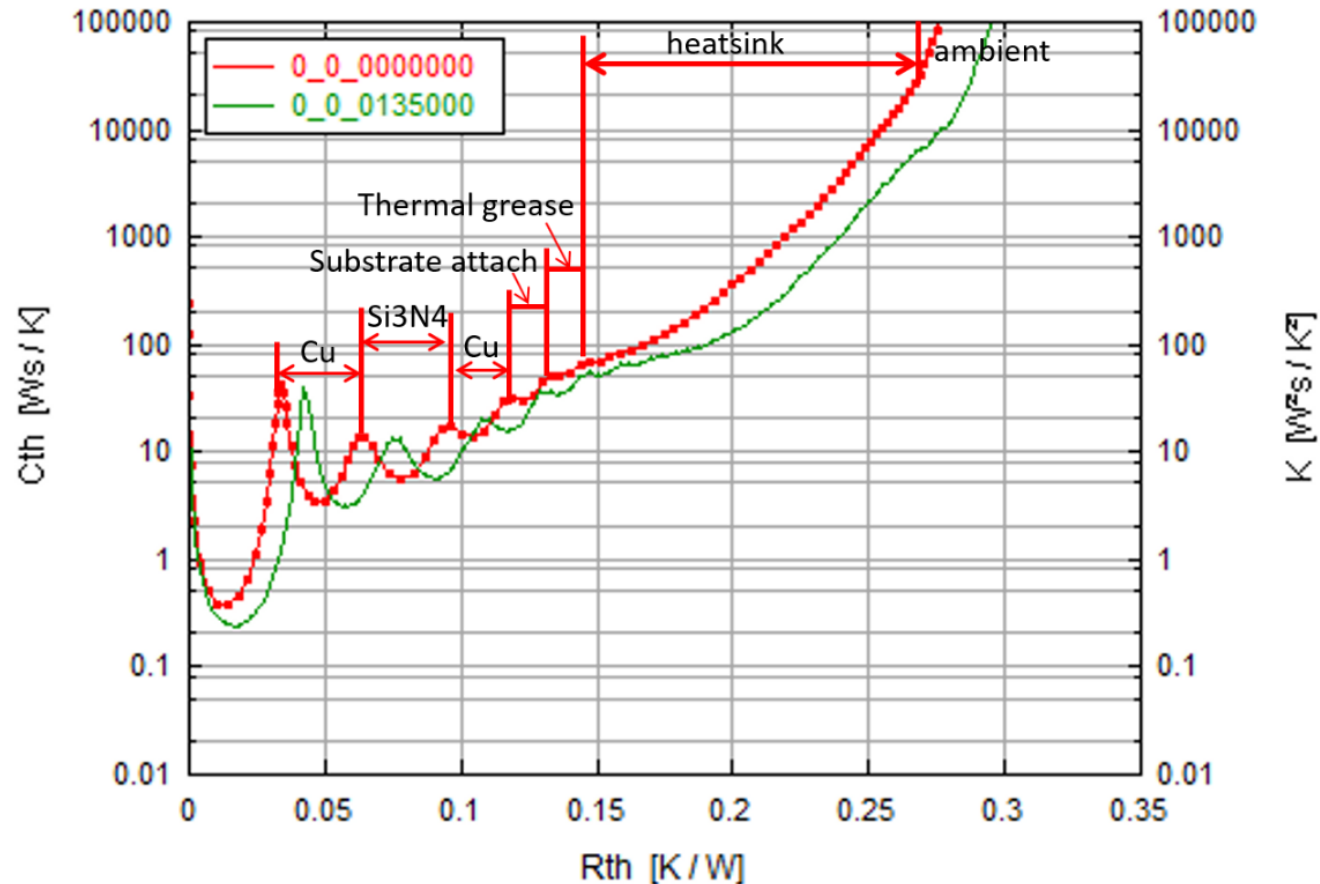
# Power modules for MEA: packaging



- clearance and creepage for high altitude
  - IEC 60664 derating factors x3.02 for 10,000m

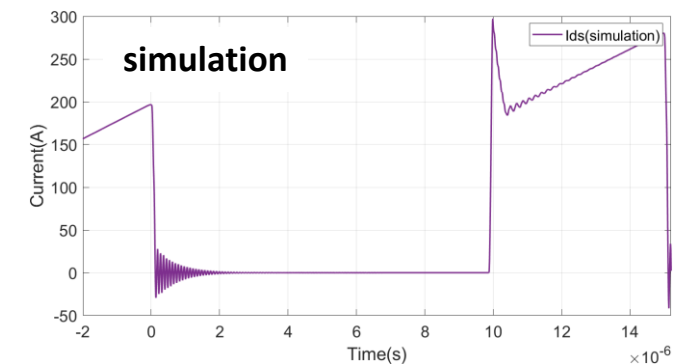
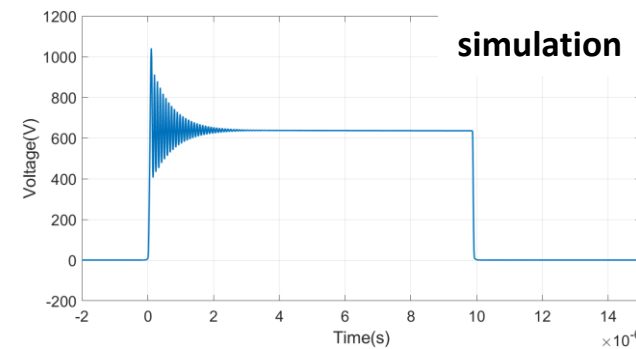
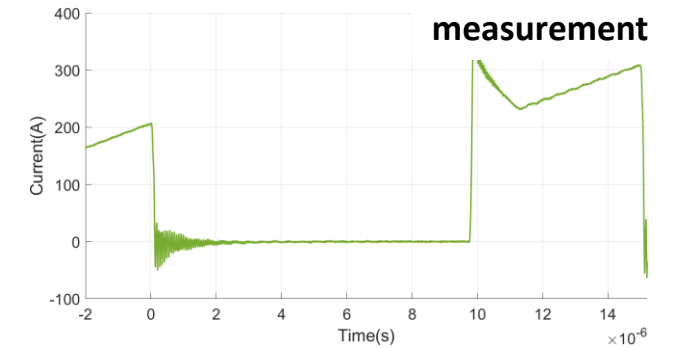
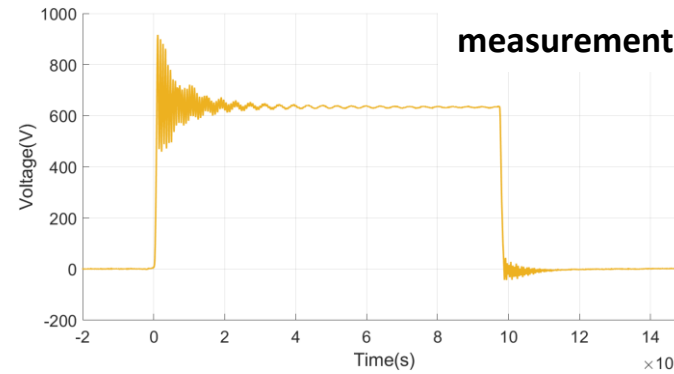
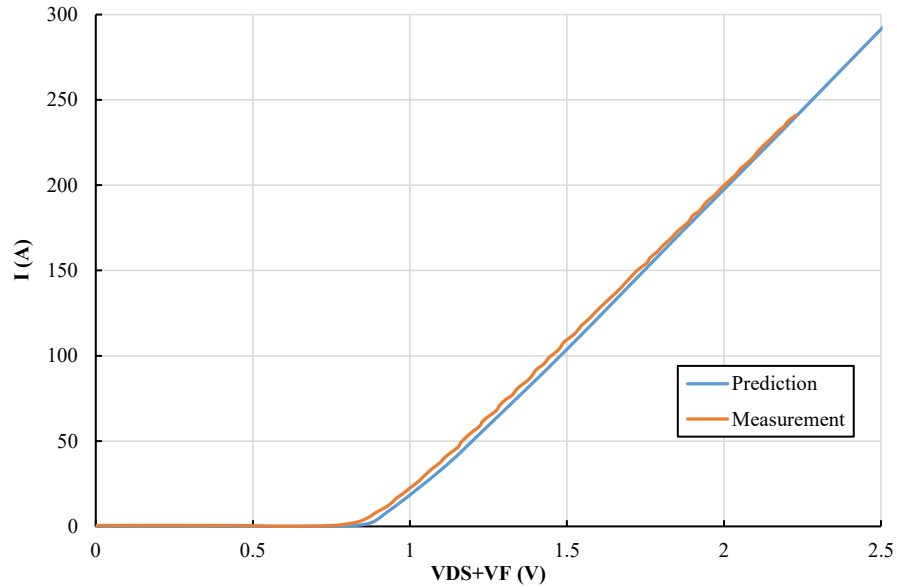
final power module design

# Power modules for MEA: reliability



- High reliability demonstrated
- >135000 power cycles
- >100 cycles of thermal shock
- No obvious changes in the output characteristics of dies after 100cycles

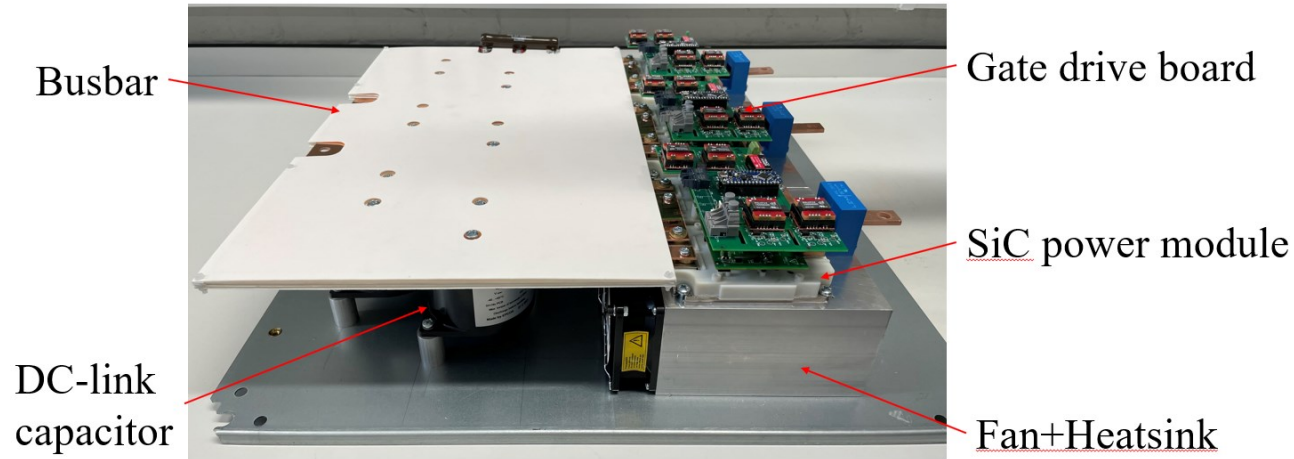
# Power modules for MEA: measurements



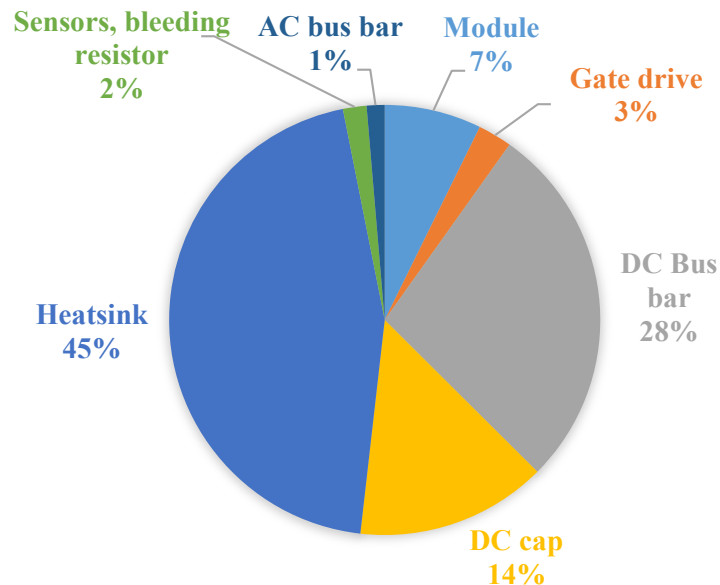
Models predict both static and dynamic characteristics accurately

	Oscillation frequency (MHz)	Maximum voltage (V)
T1(experiment)	15.8	807
T1(simulation)	15.6	812
T2(experiment)	12.7	920
T2(simulation)	11.2	1020
T3(experiment)	13.0	960
T3(simulation)	11.5	1063
T4(experiment)	16.4	779
T4(simulation)	17.2	785



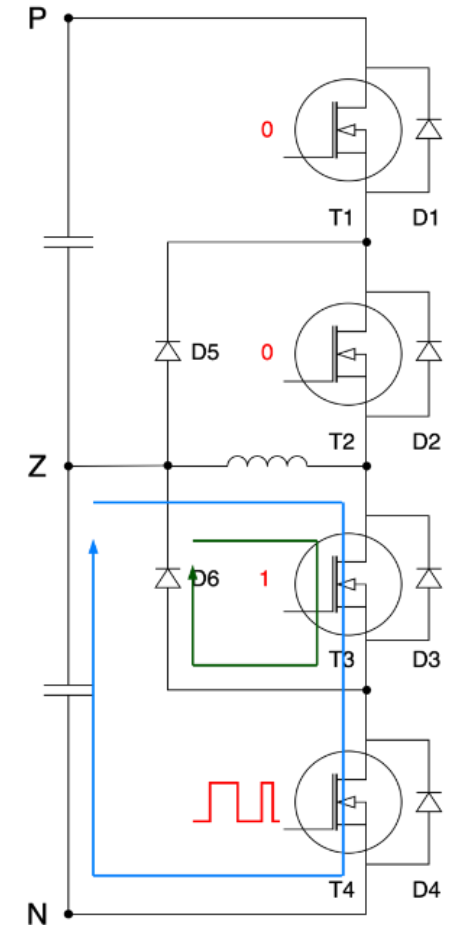
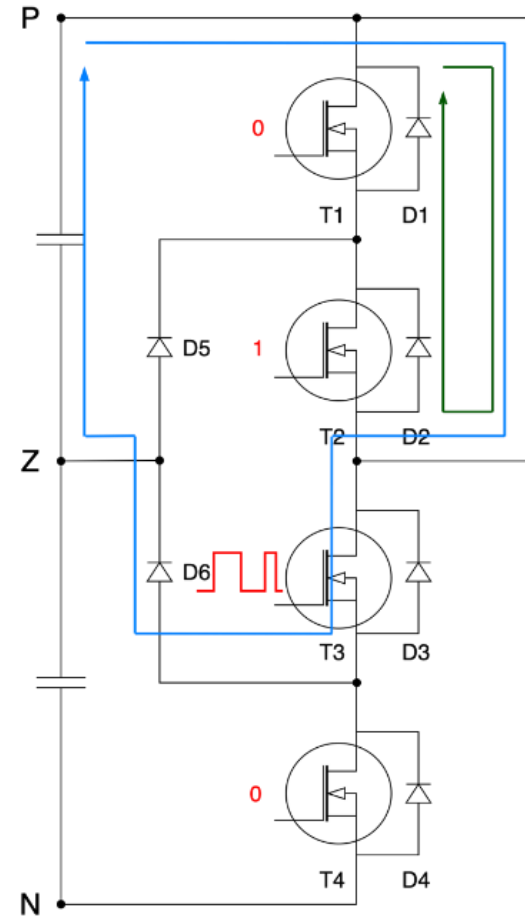
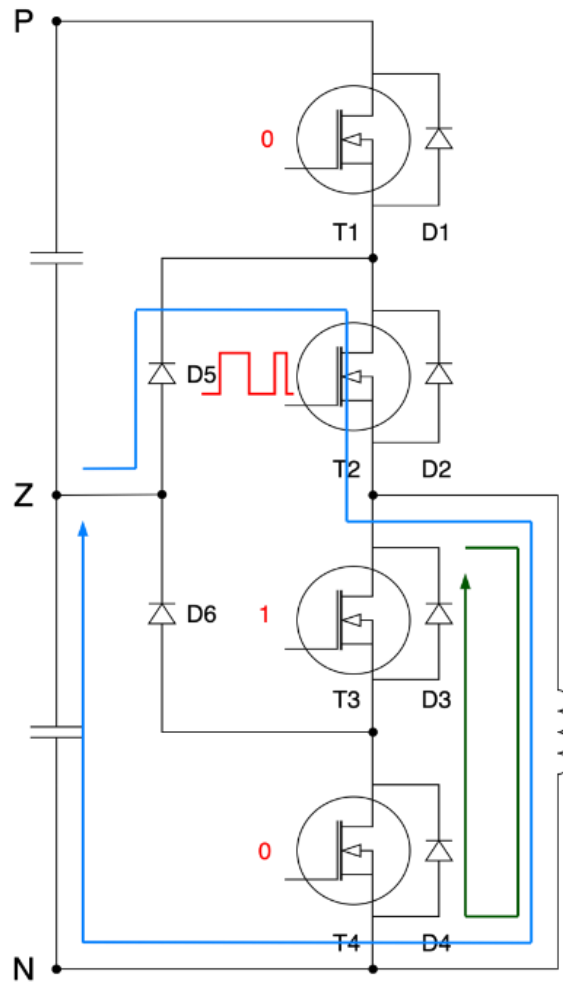
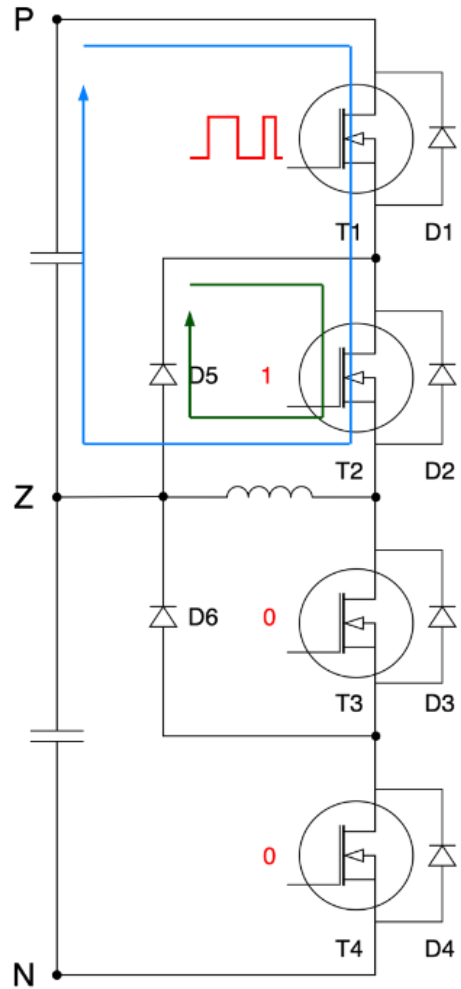


- Total weight 18.6kg.
- Assuming a nominal power of 350kW → power density of 18.8kW/kg.
- Over 45% of the total mass is due to the cooling system and a further 42% due to DC busbars and Caps.
  - Since these were not the main focus of the project, no particular optimisation of the cooling system and DC link has been performed



Weight breakdown of the converter prototype (the total weight is 18.6kg, excluding the enclosure and cables).

# Power modules for MEA: converter tests



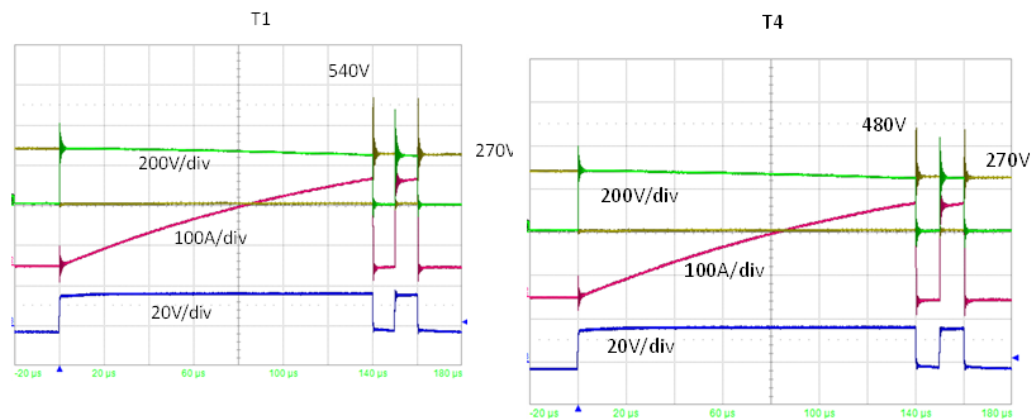


Fig. 20 Double pulse test for the outer switches T1 (left) and T4 (right)  
 Yellow:  $V_{ds}$  of switch, Green: voltage of complimentary diode, Purple: drain current, Blue:  $V_{gs}$

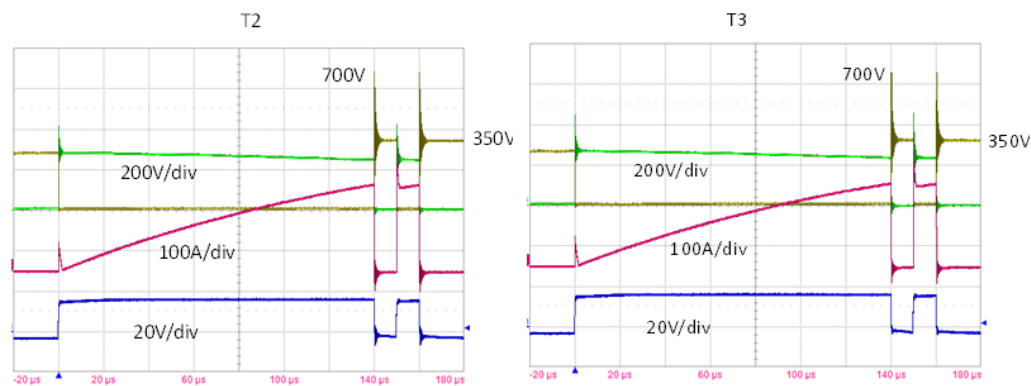
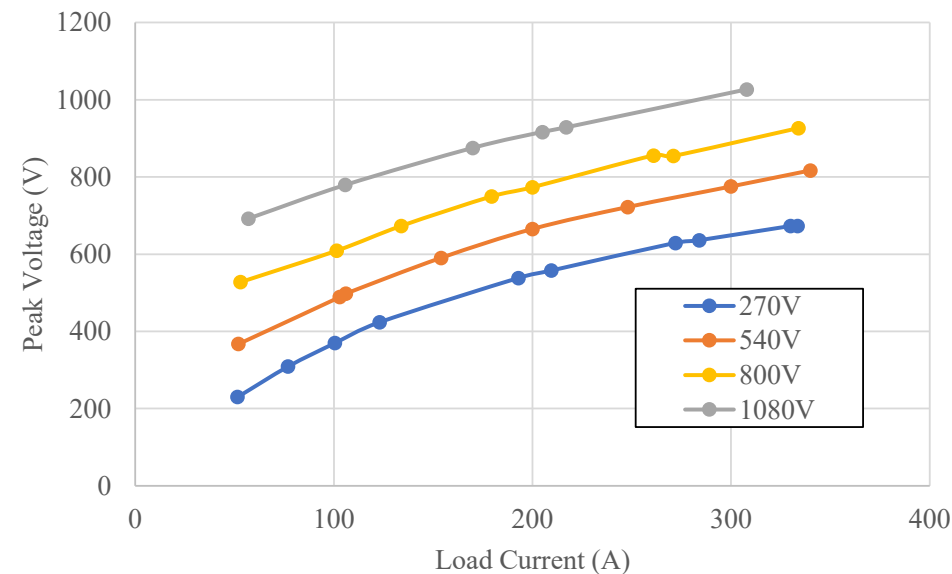


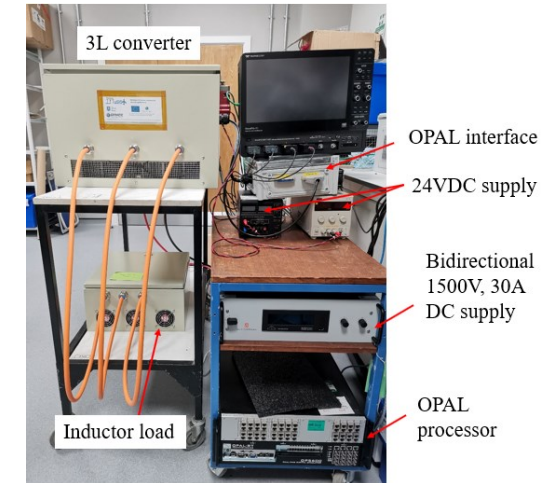
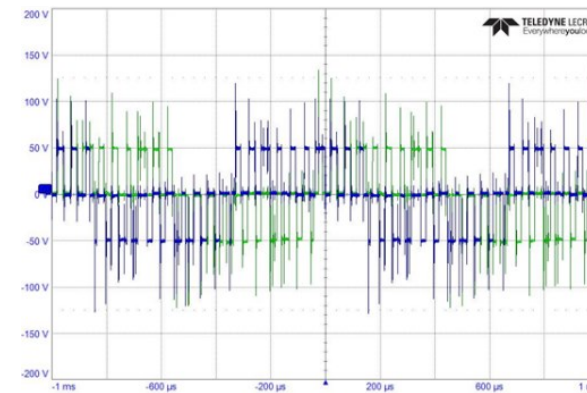
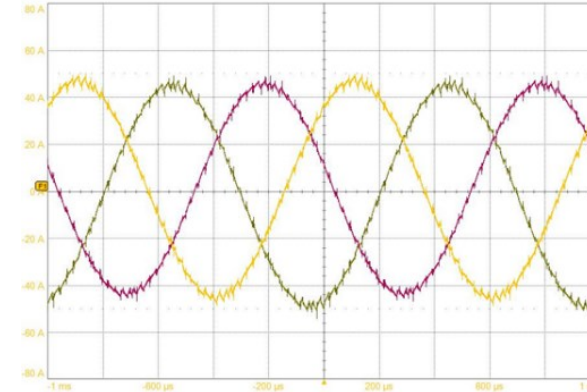
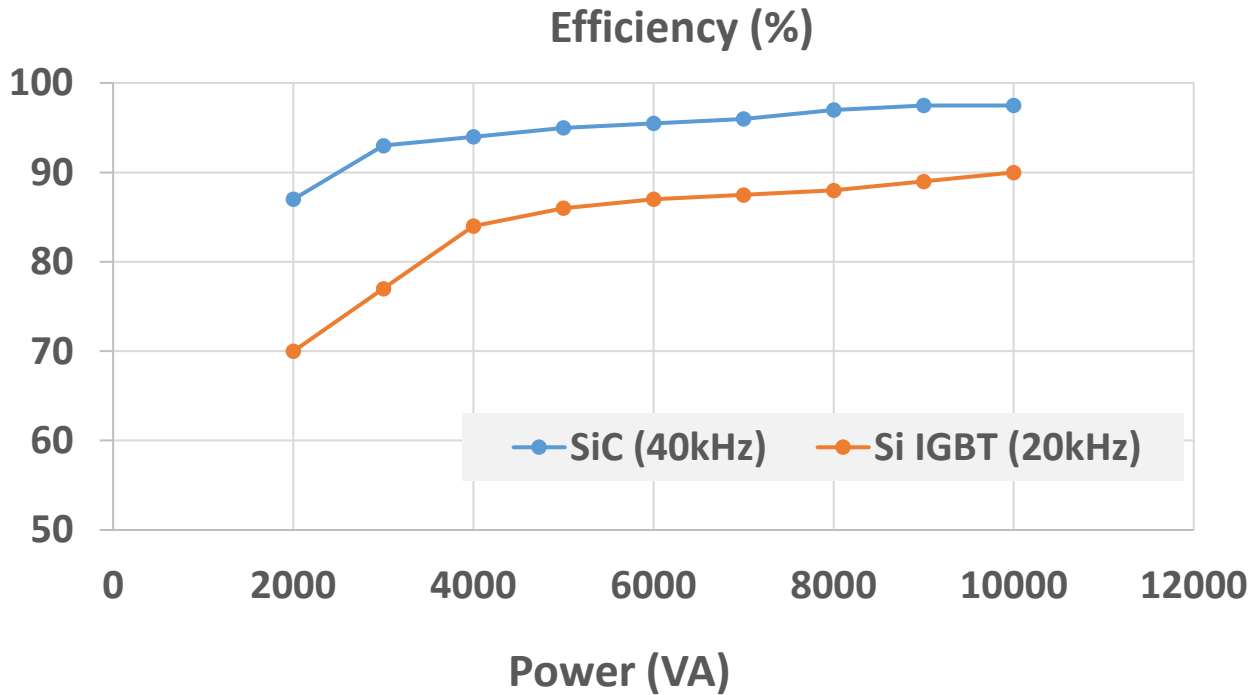
Fig. 21 Double pulse test for the inner switches T2 (left) and T3 (right)  
 Yellow:  $V_{ds}$  of switch, Green: voltage of complimentary diode, Purple: drain current, Blue:  $V_{gs}$



**SWITCHING LOSSES**

Devices	800V/300A		1080V/200A	
	Turn-on	Turn-off	Turn-on	Turn-off
T1	1.6mJ	4.5mJ	2.6mJ	3.9mJ
T2	1.3mJ	5.3mJ	1.7mJ	4.3mJ
T3	1.4mJ	4.6mJ	1.3mJ	4.0mJ
T4	1.5mJ	3.3mJ	2.9mJ	3.4mJ

# Power modules for MEA: converter tests



- WBG Power electronics will be crucial for electrification of aviation and enabling the MEA/AEA
- Analyses of optimal topologies and devices selection require detailed Multiphysics tools including electrical and thermal analysis
- Multi-level topologies will be essential in future MEA power systems as the voltages increase beyond 270V
- First 400A, 2.6mΩ, three-level NPC fully SiC module demonstrated

Thanks to:

Zixiao Li, Shangjian Dai, **University of Sheffield**

Jingru Dai, Yangang Wang, Anne Harry, **Dynex Semiconductors**

